



## **USER'S MANUAL**

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### **RT•STAR (REAL -TIME SATELLITE TIMING AND RANGING)**

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**P/N 220-600944-40X**

**Supersedes Publication No. 1220-GEN-0101  
Dated June 4, 1998**

**CMC Electronics Inc.**  
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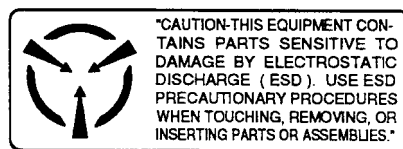
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### ELECTROSTATIC DISCHARGE

This equipment contains components which are sensitive to damage by electrostatic discharge (ESD).

Modules containing components sensitive to ESD are identified on the module by a label bearing the following marking.



When these modules have to be replaced and returned for service the following precautions should be observed:

1. Handle the modules as little as possible. Do not touch the leads, pin or tracks while handling.
2. Keep spare modules in the ESD protective packing until ready for use.
3. Discharge static before handling modules (removal or replacement) by touching a grounded metallic surface such as rack or cabinet hardware. Use of wrist strap grounded through a one megohm resistor is preferred when handling modules. (This ground should be the same as the equipment ground).
4. Do not slide static-sensitive modules over any surface.
5. Clothing must not come in contact with components or assemblies. Short sleeves are preferred; if long sleeves are worn then should be rolled up.
6. Package parts properly for storage or transportation. Modules which are removed from the equipment should be placed into ESD protective packing immediately. Do not place any paper, card or other plastic inside the ESD protective packing.
7. When packing these modules for storage or transportation, keep them in the bag. Fold over and seal the mouth of the bag to keep out any static generating packing material (eg, foamed polystyrene). Pack around the bag firmly to prevent motion which could generate static.

## WARRANTY

In the case of any ESD sensitive module bearing the marking described above which is received by CMC not in ESD protective packing, other than the initially reported fault, all warranty, present or future, is voided for failure related to ESD sensitive components.

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## GLOSSARY OF TERMS

**ASCII** - A 7 bit wide serial code describing numbers, upper and lower case alpha characters, special and non-printing characters.

**Address field** - for sentences in the NMEA standard, the fixed length field following the beginning sentence delimiter "\$" (HEX 24). For NMEA approved sentences, composed of a two character talker identifier and a three character sentence formatter. For proprietary sentences, composed of the character "P" (HEX 50) followed by a three character manufacturer identification code.

**Almanac** - a set of orbit parameters that allows calculation of approximate GPS satellite positions and velocities. The almanac is used by a GPS receiver to determine satellite visibility and as an aid during acquisition of GPS satellite signals.

**Attenuation** - reduction of signal strength.

**Azimuth** - the horizontal direction of a celestial point from a terrestrial point, expressed as the angular distance from 000° (reference) clockwise through 360°. The reference point is generally True North, but may be Magnetic North, or Relative (ship's head).

**Bearing** - the horizontal direction of one terrestrial point from another terrestrial point, expressed as the angular distance from a reference direction, usually measured from 000° at the reference direction clockwise through 360°. The reference point may be True North, Magnetic North, or Relative (ship's head).

**Carrier** - the steady transmitted RF signal whose amplitude, frequency, or phase may be modulated to carry information.

**Checksum** - by NMEA standard, a validity check performed on the data contained in the sentences, calculated by the talker, appended to the message, then recalculated by the listener for comparison to determine if the message was received correctly. Required for some sentences, optional for all others.

**Circular Error Probable (CEP)** - the radius of a circle, centered at the user's true location, that contains 50 percent of the individual position measurements made using a particular navigation system.

**Coarse Acquisition (C/A) Code** - a spread spectrum direct sequence code that is used primarily by commercial GPS receivers to determine the range to the transmitting GPS satellite. Uses a chip rate of 1.023 MHz.

**Communication protocol** - a method established for message transfer between a talker and a listener which includes the message format and the sequence in which the messages are to be transferred. Also includes the signalling requirements such a baud rate, stop bits, parity, and bits per character.

**Control segment** - the Master Control Station and the globally dispersed Monitor Stations used to manage the GPS satellites, determine their precise orbital parameters, and synchronize their clocks.

**Course** - the horizontal direction in which a vessel is to be steered or is being steered; the direction of travel through the air or water. Expressed as angular distance from reference North (either true, magnetic, compass, or grid), usually 000° (north), clockwise through 360°. Strictly, the term applies to direction through the air or water, not the direction intended to be made good over the ground (see *track*). Differs from heading.

**Cycle slip** - an error in the continuous count of carrier phase cycles.

**Dead Reckoning (DR)** - the process of determining a vessel's approximate position by applying from its last known position a vector or a series of consecutive vectors representing the run that has since been made, using only the courses being steered, and the distance run as determined by log, engine rpm, or calculations from speed measurements.

**Destination** - the immediate geographic point of interest to which a vessel is navigating. It may be the next waypoint along a route of waypoints or the final destination of a voyage.

**Differential GPS (DGPS)** - a technique to improve GPS accuracy that uses pseudorange errors measured at a known location to improve the measurements made by other GPS receivers within the same general geographic area.

**Dilution of Precision (DOP)** - A numerical value expressing the confidence factor of the position solution based on current satellite geometry. The lower the value, the greater the confidence in the solution. DOP can be expressed in the following forms:

GDOP	- all parameters are uncertain (latitude, longitude, height, clock offset)
PDOP	- 3D parameters are uncertain (latitude, longitude, height)
HTDOP	- 2D parameters and time are uncertain (latitude, longitude, time)
HDOP	- 2D parameters are uncertain (latitude, longitude)
VDOP	- height is uncertain
TDOP	- clock offset is uncertain

**Doppler** - the change in frequency of sound, light or other wave caused by movement of its source relative to the observer.

**Doppler aiding** - a signal processing strategy, which uses a measured Doppler shift to help a receiver smoothly track the GPS signal, to allow more precise velocity and position measurement.

**Earth-Centered-Earth-Fixed (ECEF)** - a right-hand Cartesian coordinate system with its origin located at the center of the Earth. The coordinate system used by GPS to describe three-dimensional location.

**ECEF** - Earth-Centered-Earth-Fixed coordinates are centered on the WGS-84 reference ellipsoid, have the "Z" axis aligned with the Earth's spin axis, the "X" axis through the intersection of the Prime Meridian and the Equator and the "Y" axis is rotated 90 degrees East of the "X" axis about the "Z" axis.

**Ephemeris** - a set of satellite orbit parameters that is used by a GPS receiver to calculate precise GPS satellite positions and velocities. The ephemeris is used in the determination of the navigation solution and is updated periodically by the satellite to maintain the accuracy of GPS receivers.

**Field** - a character or string of characters immediately preceded by a field delimiter.

**Fixed field** - a field in which the number of characters is fixed. For data fields, such fields are shown in the sentence definitions with no decimal point. Other fields which fall into this category are the address field and the checksum field (if present).

**Flash ROM** - Programmable read-only memory.

**GDOP** - Geometric Dilution of Precision - A numerical value expressing the confidence factor of the position solution based on current satellite geometry. Assumes that 3D position (latitude, longitude, height) and receiver clock offset (time) are variables in the solution. The lower the GDOP value, the greater the confidence in the solution.

**Geodetic datum** - the reference ellipsoid surface that defines the coordinate system.

**Geoid** - the figure of the earth considered as a sea level surface extended continuously through the continents. The actual geoid is an equipotential surface coincident with mean sea level to which at every point the plumb line (direction in which gravity acts) is perpendicular.

**Geostationary** - a satellite orbit along the equator that results in a constant fixed position over a particular reference point on the earth's surface. (GPS satellites are not geostationary.)

**Global Positioning System (GPS)** - full name NAVSTAR Global Positioning System, a space-based radio positioning system which provides suitably equipped users with accurate position, velocity and time data. When fully operational, GPS will provide this data free of direct user charge worldwide, continuously, and under all weather conditions. The GPS constellation will consist of 24 orbiting satellites, four equally spaced around each of six different orbital planes. The system is being developed by the Department of Defense under U.S. Air Force management.

**Great circle** - the shortest distance between any two points along the surface of a sphere or ellipsoid, and therefore the shortest navigation distance between any two points on the Earth. Also called Geodesic Line.

**HDOP** - Horizontal Dilution of Precision - A numerical value expressing the confidence factor of the horizontal position solution based on current satellite geometry. Makes no constraint assumptions about time, and about height only if the FIX HEIGHT command has been invoked. The lower the HDOP value, the greater the confidence in the solution.

**HTDOP** - Horizontal position and Time Dilution of Precision - A numerical value expressing the confidence factor of the position solution based on current satellite geometry. Assumes height is known if the FIX HEIGHT command has been invoked. If not, it will give the normalized precision of the horizontal and time parameters given that nothing has been constrained. The lower the HTDOP value, the greater the confidence factor.

**Heading** - the direction in which a vessel points or heads at any instant, expressed in degrees 000° clockwise through 360° and may be referenced to True North, Magnetic North, or Grid North. The heading of a vessel is also called the ship's head. Heading is a constantly changing value as the vessel oscillates or yaws across the course due to the effects of the air or sea, cross currents, and steering errors.

**L1 frequency** - the 1575.42 MHz GPS carrier frequency which contains the coarse acquisition (C/A) code, as well as encrypted P-code, and navigation messages used by commercial GPS receivers.

**L2 frequency** - a secondary GPS carrier, containing only encrypted P-code, used primarily to calculate signal delays caused by the ionosphere. The L2 frequency is 1227.60 MHz.

**Magnetic bearing** - bearing relative to magnetic north; compass bearing corrected for deviation.

**Magnetic heading** - heading relative to magnetic north.

**Magnetic variation** - the angle between the magnetic and geographic meridians at any place, expressed in degrees and minutes east or west to indicate the direction of magnetic north from true north.

**Mask angle** - the minimum GPS satellite elevation angle permitted by a particular GPS receiver design. Satellites below this angle will not be used in position solution.

**Measurement error variance** - the square of the standard deviation of a measurement quantity. The standard deviation is representative of the error typically expected in a measured value of that quantity.

**Multipath errors** - GPS positioning errors caused by the interaction of the GPS satellite signal and its reflections.

**Nanosecond** -  $1 \times 10^{-9}$  second.

**Nautical mile** - any of various units of distance for sea and air navigation; in the U.S. since 1959, an international unit of linear measure equal to 1 minute of arc of a great circle of the Earth, 1,852 metres (6,076 feet).

**Null field** - by NMEA standard, indicates that data is not available for the field. Indicated by two ASCII commas, i.e., "," (HEX 2C2C), or, for the last data field in a sentence, one comma followed by either the checksum delimiter "" (HEX 2A) or the sentence delimiters <CR><LF> (HEX 0D0A). [Note: the ASCII Null character (HEX 00) is not to be used for null fields.]

**Obscuration** - term used to describe periods of time when a GPS receiver's line-of-sight to GPS satellites is blocked by natural or man-made objects.

**Origin waypoint** - the starting point of the present navigation leg, expressed in latitude and longitude.

**P-Code (precise or protected)** - a spread spectrum direct sequence code that is used primarily by military GPS receivers to determine the range to the transmitting GPS satellite. Uses a chipping rate of 10.23 MHz.

**PDOP** - Position Dilution of Precision - A numerical value expressing the confidence factor of the position solution based on current satellite geometry. 3D position (latitude, longitude, height) is unknown. The lower the PDOP value, the greater the confidence factor.

**PRN** - Pseudo-Random Noise number - the identify of the GPS satellites as determined by a GPS receiver. Since all GPS satellites must transmit on the same frequency, they are distinguished by their pseudo-random noise codes.

**Parallel receiver** - a receiver that monitors four or more satellites simultaneously with independent channels.

**Precise Positioning Service (PPS)** - the GPS positioning, velocity, and time service which will be available on a continuous, worldwide basis to users authorized by the U.S. Department of Defense (typically using P-Code).

**Pseudolite** - an Earth-based transmitter designed to mimic a satellite. May be used to transmit differential corrections.

**Pseudorange** - the calculated range from the GPS receiver to the satellite determined by taking the difference between the measured satellite transmit time and the receiver time of measurement, and multiplying by the speed of light. This measurement generally contains a large receiver clock offset error.

**Receiver channels** - a GPS receiver specification which indicates the number of independent hardware signal processing channels included in the receiver design.

**Relative bearing** - bearing relative to heading or to the vessel.

**Residual** - in the context of measurements, the residual is the misclosure between the calculated measurements, using the position solution and actual measurements.

**Route** - a planned course of travel, usually composed of more than one navigation leg.

**Satellite elevation** - the angle of the satellite above the horizon.

**Selected waypoint** - the waypoint currently selected to be the point toward which the vessel is travelling. Also called “**to**” **waypoint**, **destination** or **destination waypoint**.

**Selective Availability (SA)** - the method used by the United States Department of Defense to control access to the full accuracy achievable by civilian GPS equipment (generally by introducing timing and ephemeris errors).

**Sequential receiver** - a GPS receiver in which the number of satellite signals to be tracked exceeds the number of available hardware channels. Sequential receivers periodically reassign hardware channels to particular satellite signals in a predetermined sequence.

**Spherical Error Probable (SEP)** - the radius of a sphere, centered at the user's true location, that contains 50 percent of the individual three-dimensional position measurements made using a particular navigation system.

**Spheroid** - sometimes known as ellipsoid; a perfect mathematical figure which very closely approximates the geoid. Used as a surface of reference for geodetic surveys. The geoid, affected by local gravity disturbances, is irregular.

**Standard Positioning Service (SPS)** - a positioning service made available by the United States Department of Defense which will be available to all GPS civilian users on a continuous, worldwide basis (typically using C/A code)

**SV** - Space Vehicle ID, sometimes used as SVID; also used interchangeably with Pseudo-Random Noise Number (PRN).

**TDOP** - Time Dilution of Precision - A numerical value expressing the confidence factor of the position solution based on current satellite geometry. The lower the TDOP value, the greater the confidence factor.

**Three-dimensional coverage (hours)** - the number of hours-per-day when four or more satellites are available with acceptable positioning geometry. Four visible satellites are required to determine location and altitude.



**Three-dimensional (3D) navigation** - navigation mode in which altitude and horizontal position are determined from satellite range measurements.

**Time-To-First-Fix (TTFF)** - the actual time required by a GPS receiver to achieve a position solution. This specification will vary with the operating state of the receiver, the length of time since the last position fix, the location of the last fix, and the specific receiver design.

**Track made good** - the single resultant direction from a point of departure to a point of arrival or subsequent position at any given time; may be considered synonymous with Course Made Good.

**True bearing** - bearing relative to true north; compass bearing corrected for compass error.

**True heading** - heading relative to true north.

**Two-dimensional coverage (hours)** - the number of hours-per-day with three or more satellites visible. Three visible satellites can be used to determine location if the GPS receiver is designed to accept an external altitude input.

**Two-dimensional (2D) navigation** - navigation mode in which a fixed value of altitude is used for one or more position calculations while horizontal (2D) position can vary freely based on satellite range measurements.

**Undulation** - the distance of the geoid above (positive) or below (negative) the mathematical reference ellipsoid (spheroid). Also known as geoidal separation, geoidal undulation, geoidal height.

**Universal Time Coordinated (UTC)** - this time system uses the second-defined true angular rotation of the Earth measured as if the Earth rotated about its Conventional Terrestrial Pole. However, UTC is adjusted only in increments of one second. The time zone of UTC is that of Greenwich Mean Time (GMT).

**Update rate** - the GPS receiver specification which indicates the solution rate provided by the receiver when operating normally.

**VDOP** - Vertical Dilution of Precision - A numerical value expressing the confidence factor of the position solution based on current satellite geometry. The lower the VDOP value, the greater the confidence factor.

**Variable field** - by NMEA standards, a data field which may or may not contain a decimal point and which may vary in precision following the decimal point depending on the requirements and the accuracy of the measuring device.

**WGS-84** - World Geodetic System 1984 is an ellipsoid designed to fit the shape of the entire Earth as well as possible with a single ellipsoid. It is often used as a reference on a worldwide basis, while other ellipsoids are used locally to provide a better fit to the Earth in a local region. GPS uses the center of the WGS-84 ellipsoid as the center of the GPS ECEF reference frame.

**Waypoint** - a reference point on a track.

## GLOSSARY OF ACRONYMS

1PPS	One Pulse Per Second
2D	Two Dimensional
3D	Three Dimensional
A/D	Analog-to-Digital
ASCII	American Standard Code for Information Interchange
BIT	Built-In Test
bps	Bits per Second
C/A Code	Coarse/Acquisition Code
CEP	Circular Error Probable
CMC	CMC Electronics Inc.
CPU	Central Processing Unit
CR	Carriage Return
CRC	Cyclic Redundancy Check
CTS	Clear To Send
dB	Decibel
DGNSS	Differential Global Navigation Satellite System
DGPS	Differential Global Positioning System
DOP	Dilution Of Precision
DSP	Digital Signal Processor
DSR	Data Set Ready
DTR	Data Terminal Ready
ECEF	Earth-Centered-Earth-Fixed
ESD	Electrostatic Discharge
FOM	Figure of Merit
GDOP	Geometric Dilution Of Precision
GMT	Greenwich Mean Time
GND	Ground
GPS	Global Positioning System
HDOP	Horizontal Dilution Of Precision
hex	Hexadecimal
HTDOP	Horizontal position and Time Dilution Of Precision
Hz	Hertz
IC	Integrated Circuit
IF	Intermediate Frequency
I/O	Input/Output
IODE	Issue of Data (Ephemeris)
IRQ	Interrupt Request

LF	Line Feed
LHCP	Left Hand Circular Polarization
LNA	Low Noise Amplifier
LO	Local Oscillator
lsb	Least significant bit
msb	Most significant bit
msec	millisecond
MSL	Mean sea level
MTBF	Mean Time Between Failures
N.mi.	Nautical mile
NCO	Numerically Controlled Oscillator
NMEA	National Marine Electronics Association
nsec	nanosecond
OCXO	Oven Controlled Crystal Oscillator
OEM	Original Equipment Manufacturer
PC	Personal Computer
PCB	Printed Circuit Board
P Code	Precise Code
PDOP	Position Dilution Of Precision
PLL	Phase Lock Loop
PPS	Precise Positioning Service or Pulse Per Second
PRN	Pseudo-Random Noise number
PVT	Position Velocity Time
RAM	Random Access Memory
RF	Radio Frequency
RHCP	Right Hand Circular Polarization
ROM	Read Only Memory
RTC	Real-Time Clock
RTCA	Radio Technical Commission for Aviation Services
RTCM	Radio Technical Commission for Maritime Services
RTK	Real Time Kinematic
RTS	Request To Send
RXD	Received Data
SA	Selective Availability
SEP	Spherical Error Probable
SNR	Signal-to-Noise Ratio
SPS	Standard Positioning Service
SRAM	Static Random Access Memory
SV	Space Vehicle

TCXO	Temperature Compensated Crystal Oscillator
TDOP	Time Dilution Of Precision
TTFF	Time-To-First-Fix
TXD	Transmitted Data
UART	Universal Asynchronous Receiver Transmitter
UDRE	User Differential Range Error
UTC	Universal Time Coordinated
VDOP	Vertical Dilution of Precision
VSWR	Voltage Standing Wave Ratio
WGS	World Geodetic System
wpt	Waypoint
XTE	Crosstrack Error

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## SECTION I - INTRODUCTION

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## SECTION I

### INTRODUCTION

#### PURPOSE OF THE MANUAL

This user manual defines the design, operational characteristics, physical, interface, functional and performance requirements for the receiver along with the installation and operation procedures.

#### SYSTEM OVERVIEW

The receiver is a Global Positioning System (GPS) Standard Positioning Service (SPS) single board twelve-channel Real-Time Kinematic (RTK) receiver for embedding in Original Equipment Manufacturer (OEM) consumer market navigation systems.

Code Differential GPS (Code DGPS) is the regular Global Positioning System (GPS) with an additional correction (differential) signal added. This correction signal improves the accuracy of GPS and can be broadcast over any authorized communication channel.

The GPS determined position of a base station is computed and compared to its surveyed geodetic position. The differential information is transmitted to user receivers by radio or other means. These differences can then be matched up with GPS measurements from the roving GPS receiver, and used to remove the systematic (correctable) error factors.

The differential correction technique described above applies to GPS receivers performing code-phase navigation, which use the transmitted GPS code information to compute pseudo-ranges (distances) from the earth to the GPS satellites in space. When a receiver navigates in carrier-phase mode, it is measuring a different GPS observable, namely the GPS carrier wave. In order to obtain high accuracy with carrier-phase measurements, it is necessary for a roving GPS receiver to use information (i.e. carrier-phase measurements) from a base receiver to compute the integer number of GPS wavelengths between the roving GPS receiver's antenna and the satellite(s). This technique yields accuracies in the cm-range in dynamic environments and is called Real-time Kinematic, or RTK, GPS.

A DGPS system therefore consists of at least two units: a base station and one or several roving units. The base station broadcasts its differential data and the roving units receive it through a data port, directly connected to a radio receiver. The roving units can then display velocity, time and other information as needed for their marine, terrestrial, or aeronautical applications.

The receiver with a separate GPS antenna, decodes the GPS satellites RF signal and interfaces with a host system to provide three dimensional user position and velocity, time and other status information at a maximum rate of once per second. The receiver uses WGS-84 as its geographic reference.

The receiver has 12 independent parallel channels each capable of simultaneously tracking a GPS satellite signal. The receiver makes provisions for external initialization of data to support faster GPS signal acquisition. The receiver also has the capability to operate in differential mode where inherent GPS inaccuracies can be reduced through the utilization of differential corrections and measurements. Figure 1-1 illustrates the receiver single board.



Figure 1-1. Receiver Single Board

The main features are listed as follows:

- Nominal position accuracies of 20 cm (CEP) in real-time.
- Configurable either as a roving unit or a base station.
- Twelve channel correlator for all-in-view satellite tracking.
- Single chip RF Front End.
- Supports active and passive antennas.
- Single 5V input operation.
- Complete GPS receiver and navigator on a single compact board.
- Operation under standard temperature range (-30°C to +75°C). Optional extended temperature range (-40°C to +85°C).
- 1 PPS Output aligned on GPS Time  $\pm 200$  ns
- 1,2,5 or 10 Hz Measurement Output Aligned on GPS Time
- Support for 62 predefined datums.
- Upgradeable software (stored in Flash memory) via the RS-232 serial port.
- Code and Carrier tracking of L1 GPS frequency for increased accuracy.
- Support of vehicle dynamics up to 4g.
- Retention of satellite almanac and ephemeris data in non-volatile memory for rapid time-to-first-fix (TTFF) after power interruption.
- Very fast signal reacquisition due to signal masking (obstruction or vehicle attitude).
- Two serial input/output data ports. One for host communication, the second one for differential data input or other customer needs. Both can be used for the maintenance (reprogramming) mode.
- On-board rechargeable lithium battery (optional).

Custom Application Optional Features:

- Spare CPU time.
- Third serial input/output data port.
- Memory expansion: FLASH, EEPROM and SRAM memories.
- 2 Hz and 5 Hz PVT Output (Optional)

The receiver is available in 3 formats:

- as an OEM board
- within the Development Kit
- within the STARBOX casing

The Development Kit is an equipment set permitting easy evaluation of the receiver. A full description of this kit is provided in Appendix A.

The STARBOX casing is a special packaging of the receiver. A full description of the STARBOX is provided in Appendix B.



**RELATED PUBLICATIONS**

The related publications are listed in Figure 1-2.

PUBLICATION NAME	PUBLICATION NAME
[1] ICD-GPS-200 Rev. B	NAVSTAR GPS Space Segment/Navigation Interface
[2] RTCM-104 version 2.1 January 1994	Recommended Standards for Differential NAVSTAR GPS Radio Technical Commission for Maritime Services
[3] SAE J1211	Recommended Environmental Practices for Electronic Equipment Design
[4] NMEA-0183 Rev 2.20	National Marine Electronics Association Standard for Interfacing
[5] STARVIEW User's Manual	CMC #1205-GEN-0101

Figure 1-2. Related Publications

**EQUIPMENT IDENTIFICATION**

Using the receiver DGPS system (roving unit and base station) requires specific hardware equipment. The nomenclature and CMC part number or model for the required equipment are listed in Figure 1-3.

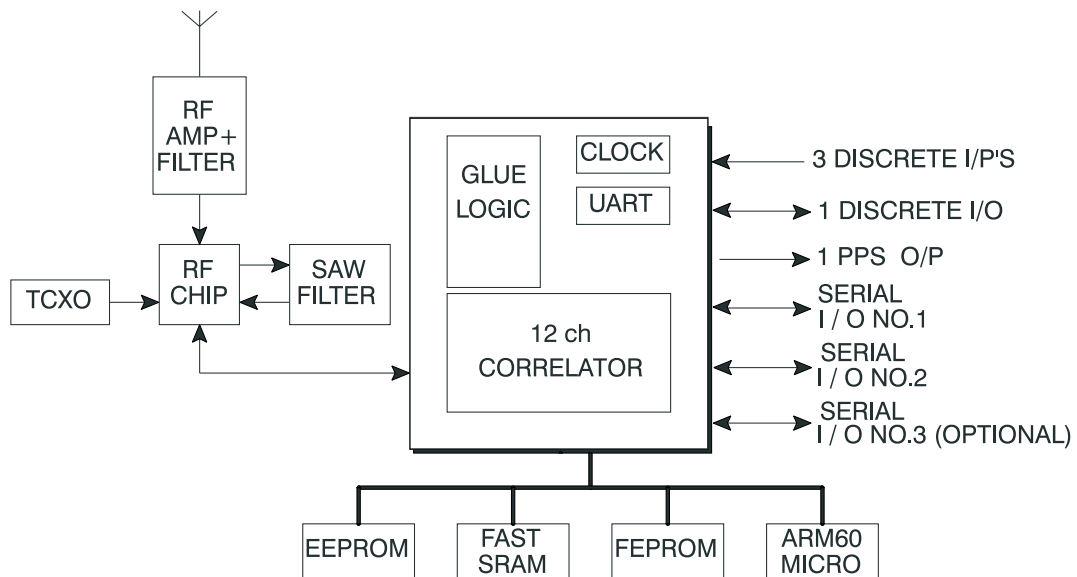
EQUIPMENT NOMENCLATURE	CMC PART NUMBER OR RECOMMENDED MODEL
Receiver Base Station	220-600944-40X
Base Station GPS Antenna	Active Choke Ring Antenna between +12dB and +36dB <sup>1</sup>
Base Station DPGS Transmission Antenna	Any UHF antenna
Base Station Transmitting Modem	GLB Model SN2TR96-450-25
Receiver Roving Unit	220-600944-40x
Roving Unit GPS Antenna	Active Geodetic Antenna between +12dB and +36dB <sup>1</sup>
Roving Unit DPGS Transmission Antenna	Any UHF unidirectional antenna
Roving Unit Receiving Modem	GLB Model SN2RX96-450

<sup>1</sup> Refer to Appendix D, or contact CMC for our list of antennas (sold separately).

Figure 1-3. Equipment Identification

## SYSTEM ARCHITECTURE

Figure 1-4 below depicts the block diagram of the receiver assembly.



G704086cdr

Figure 1-4. Receiver Block Diagram

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## SECTION II - RECEIVER SPECIFICATIONS

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## SECTION II

### RECEIVER SPECIFICATIONS

#### NAVIGATION PERFORMANCE

The position and velocity outputs meet the accuracies defined in Figure 2-1 under the dynamic conditions of 500 m/s and linear acceleration of up to  $\pm 4.0g$ . Specified accuracies are achieved with a 95% probability.

NAVIGATION ACCURACIES	SA INACTIVE	SA ACTIVE	DGPS	RTK
Receiver Performance 2 SIGMA (95%)				
Horizontal Position	30 meters	100 meters	2 meters	0.2 meters
Ground Speed*	0.13 m/s	0.3 m/s	0.05 m/s	0.05 m/s
Track Angle True**	1.0 deg	3.0 deg	0.1 deg	0.1 deg
Vertical Speed	0.16 m/s	0.6 m/s	0.1 m/s	0.1 m/s
Altitude	40 meters	160 meters	5 meters	0.2 meters
N-S Velocity*	0.1088 m/s	0.21 m/s	0.035 m/s	0.035 m/s
E-W Velocity*	0.1088 m/s	0.21 m/s	0.035 m/s	0.035 m/s
Time***	1 usec	1 usec	1 usec	1 usec

\* }

\*\* } See p. 10 of document 1826-1127, Rev. K

\*\*\* }

Figure 2-1. Position and Velocity Outputs

The accuracies are met for the following conditions:

HDOP = 1.5  
VDOP = 2.0  
TDOP = 0.8

#### A. FIGURE OF MERIT

The receiver provides an estimated accuracy level. The accuracy level estimate is provided in the horizontal and vertical Figure of Merit (FOM). The FOM reflects a 95% confidence level for the position solution accuracy estimate. The FOM accounts for all major sources of errors in the pseudo ranges of the satellites used in the position solution. The error sources which are included are selective availability, ionospheric and tropospheric errors, satellite position errors based on transmitted user range error and thermal noise.

**B. TIME-TO-FIRST-FIX (TTFF)**

The receiver shall enter Navigation mode and provide valid outputs in less than 50 seconds (95%) after completion of the self-test and all of the following initialization criteria being met:

1. Valid time ( $\pm 10$  minutes) and position data ( $\pm 100$  km) from actual position.
2. Valid almanac data (less than one year old).
3. Elevation of at least 4 satellites greater than  $5^\circ$  above horizon.
4. HDOP  $< 6$ .

The time allowed for self-test and device initialization is less than 5 seconds.

In the case where the following additional conditions are met, the TTFF is reduced to less than 30 seconds (95%):

5. The unit was in SRAM Keep-Alive mode before nominal power was re-applied.
6. The last navigation fix occurred within the last 2 hours.
7. Valid ephemeris data (age of less than 4 hours) for at least 5 satellites.

With no initialization, the time from power application to valid navigation output is less than 3 minutes typically (less than 10 minutes, 95%).

**RECEIVER PERFORMANCE**

The receiver meets the performance requirements defined below under conditions of vehicle operating speeds of up to 514 m/s (limited by Canadian & US Export Laws), acceleration of up to  $\pm 4.0g$ , jerk of up to  $2 \text{ m/s}^3$ , specified temperature range (as specified herein) and minimum carrier-to-noise ratios (as specified herein).

**1. GPS Signals**

The receiver is meant to operate using the L1 GPS signal as described in Reference [1].

**2. Reacquisition**

Reacquisition is defined as resumption of tracking and measurement processing.

There is no disruption of navigation data output when a satellite signal is lost, for reasons other than a receiver power interrupt, for a period of less than or equal to 200 milliseconds.

When a satellite signal is lost, for reasons other than a receiver power interrupt, for a period greater than 200 milliseconds but less than 5 seconds, the receiver reacquires the satellite signal within 0.3 seconds after the satellite visibility has been restored.

When a satellite signal has been lost due to signal masking, the signal is typically reacquired within 2-3 seconds after the satellite signal meets the minimum input levels. The vehicle dynamics during the masking period are assumed to be less than or equal to 0.5g acceleration and 100 m/s velocity.

When total signal masking occurs, navigation will resume within 3-5 seconds of a Navigation mode criteria being met.

### 3. Measurement Rate

The receiver is capable of 10 measurements per satellite per second. A complete navigation solution is computed every second (2 per second if in 2Hz PVT mode or 5 per second if in 5Hz PVT mode) whenever a sufficient set of measurements is acquired.

### 4. Operational Signal Level Input

The receiver will operate with a signal level input from -165 dBW to -120 dBW.

### 5. RF Input Impedance

The impedance is 50 ohms with VSWR of 2.0 : 1 or better.

### 6. Receiver Noise Figure

The receiver has the following noise figure characteristics:

Typical:	3.8 dB
Maximum:	4.8 dB

in the temperature range of -40°C to +85°C and supply voltage range  $5V \pm 5\%$ .

### 7. Acquisition Sensitivity

The receiver is capable of acquiring satellite signals with a minimum input carrier-to-noise density ratio ( $C/N_0$ ) to the correlator of 34 dB-Hz.

### 8. Tracking Sensitivity

Once a signal has been acquired, the receiver is capable of tracking satellite signals with a minimum input carrier-to-noise density ratio ( $C/N_0$ ) to the correlator of 31 dB-Hz.

### 9. Input Burn-Out Protection

The receiver is capable of withstanding a signal level not exceeding +15 dBm at L1+/- 50 MHz without damage.

### 10. Out of Band CW Signal Rejection

The receiver, in a suitable system configuration, is capable of continuous operation under interference conditions specified in Figure 5.



## PHYSICAL CHARACTERISTICS

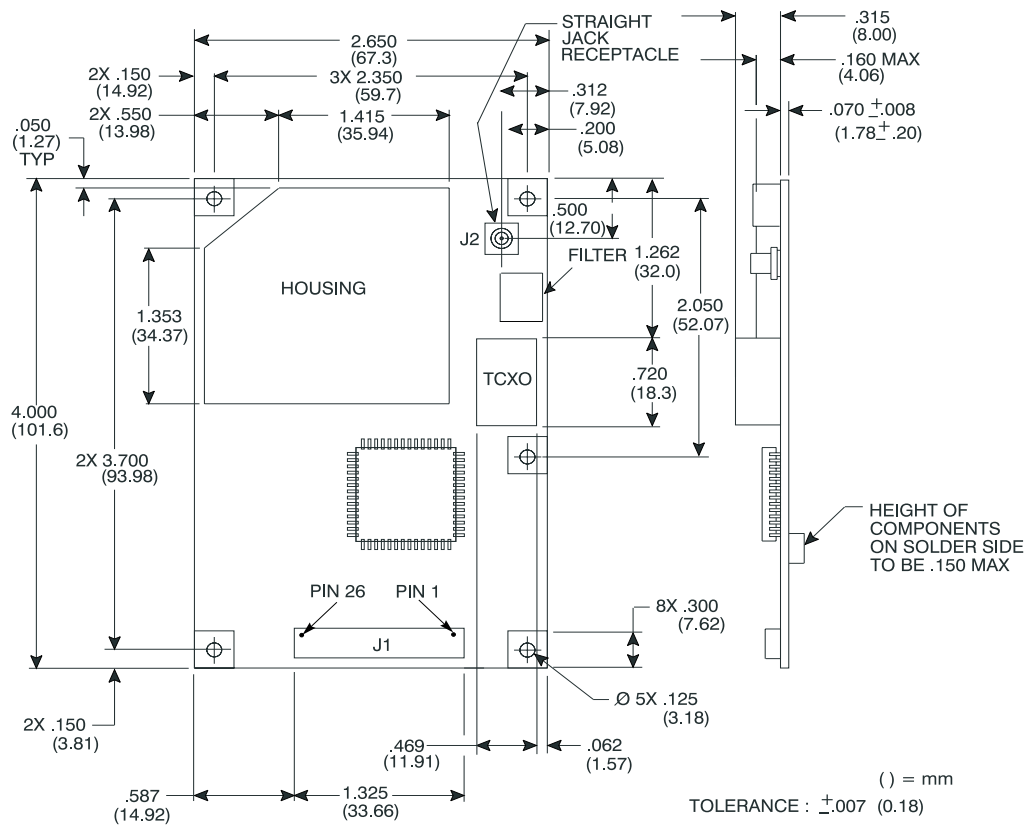
This section applies to the OEM board version of the receiver.

For details on the physical characteristics of the Development Kit version of the receiver, please refer to Appendix A.

For details on the physical characteristics of the STARBOX version of the receiver, please refer to Appendix B.

### A. OUTLINE AND FORM FACTOR

Figure 2-2 shows the OEM board outline.



G704085cdr

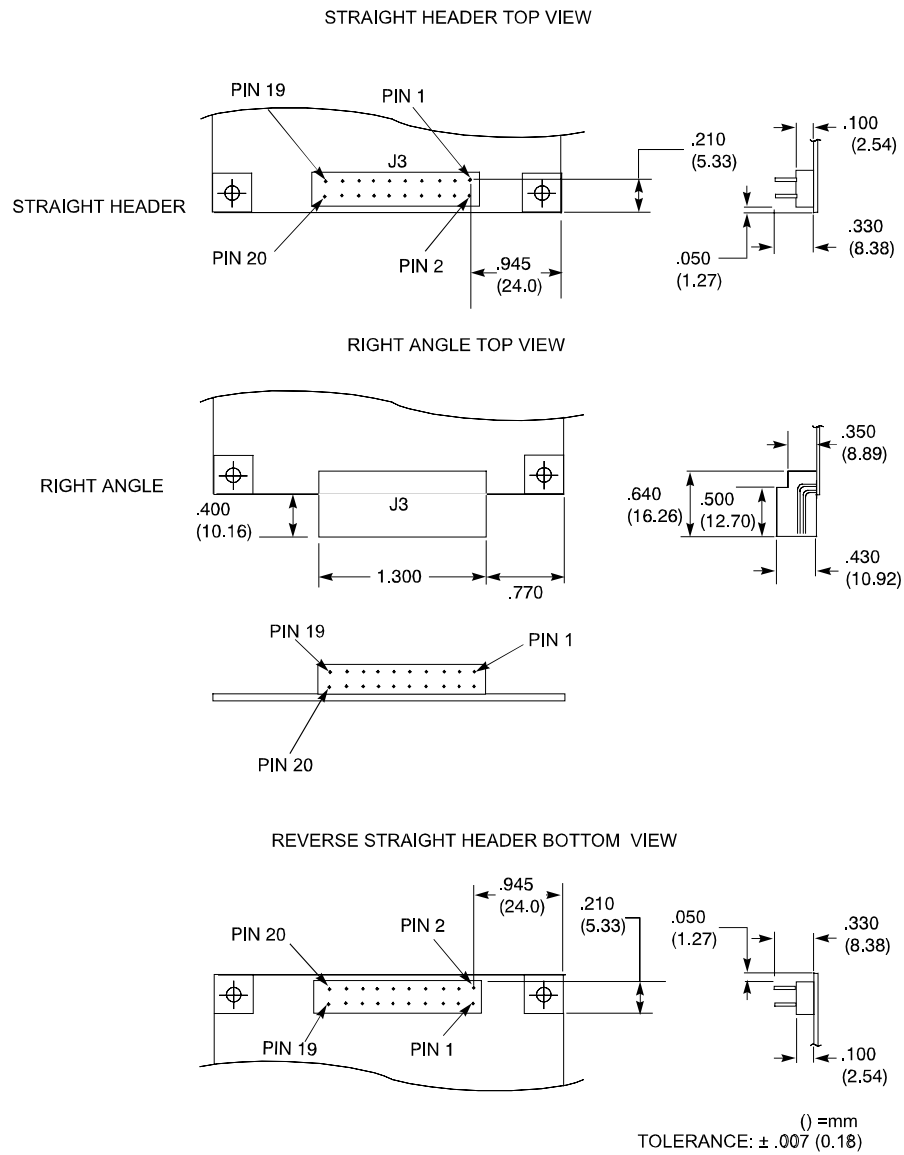
Figure 2-2. Receiver OEM Board Outline Drawing  
(VAR -401)

Figure 2-3 shows the receiver OEM board outline for:

VAR-402 : right angle connector

VAR-403 : straight header connector on top side

VAR-400 : straight header connector on bottom side



G704085acdr

Figure 2-3. Receiver OEM Board Outline Drawing  
(Var: -402, -403, -400)

## B. PACKAGING DESCRIPTION

The receiver assembly consists of one PCB, containing a shielded RF, digital and I/O sections located on both sides of the PCB and a surface mount connector. The receiver does not require heat-sinking to a metal case.

Mechanical packaging of the receiver is designed to allow for mounting within various different configurations of OEM units.

### 1. Weight Limits

The receiver weight limit is 1.76 ounces (50 grams) maximum.

### 2. Size

The receiver size is:

Height (total including components) 0.55 in. (1.4 cm)

Length 4.00 in. (10.2 cm)

Width 2.65 in. (6.7 cm)

See Figures 2-2 and 2-3 for outline drawings.

## RELIABILITY

A design goal of 55,000 hours MTBF for a Ground Fix environment is pursued through a robust design, when the receiver is installed in an OEM unit, offering reasonable environmental protection. The high reliability is ensured through concurrent engineering practices, covering all aspects of the electrical and mechanical design. Attention is paid to all features that affect the producibility, testability and maintainability of the assembly.

The MTBF calculation uses to the maximum extent possible models derived from past experience (service and test), which also account for failures due to causes other than piece-parts. When such data is not available, the analysis procedure of MIL-HDBK-217F is used, assuming a 40°C ambient temperature inside the host unit.

## ENVIRONMENTAL AND EMC REQUIREMENTS

The receiver operates within the performance requirements specified herein during and/or after exposure to the following environmental and electrical conditions. The receiver meets all specified requirements and provides performance and reliability under any natural combination of the service conditions outlined in Figure 2-3.

It shall be understood that in normal operation the environmental and EMC tests shall be performed with the receiver installed within the host unit. When in a unit the following environmental requirements of Figure 2-3 shall be met.

The basic version of the receiver dissipates 1.2W typical. The receiver relies on convection and radiation for heat dissipation. If the host unit's internal temperature is greater than the maximum operating temperature, thermal management shall provide for heat sinking of the RF shield to the host unit chassis.

Operating Temperature	-30°C to +75°C (Optional -40° C to +85°C)
Storage Temperature	-55°C to +100°C (Version without battery)
Temperature Variation	4°C per minute
Humidity	Relative Humidity up to 95%, non-condensing
Altitude	-1,000 feet to 60,000 feet (18 000 m)
Vibration operational	See SAE curve Figure 2-5
Shock	20g peak, 5 milliseconds duration (3 axes)
Dynamics	Velocity : 514 m/s
	Acceleration : 4g
	Jerk : 2 m/s <sup>3</sup>

Figure 2-4. Environmental Categories

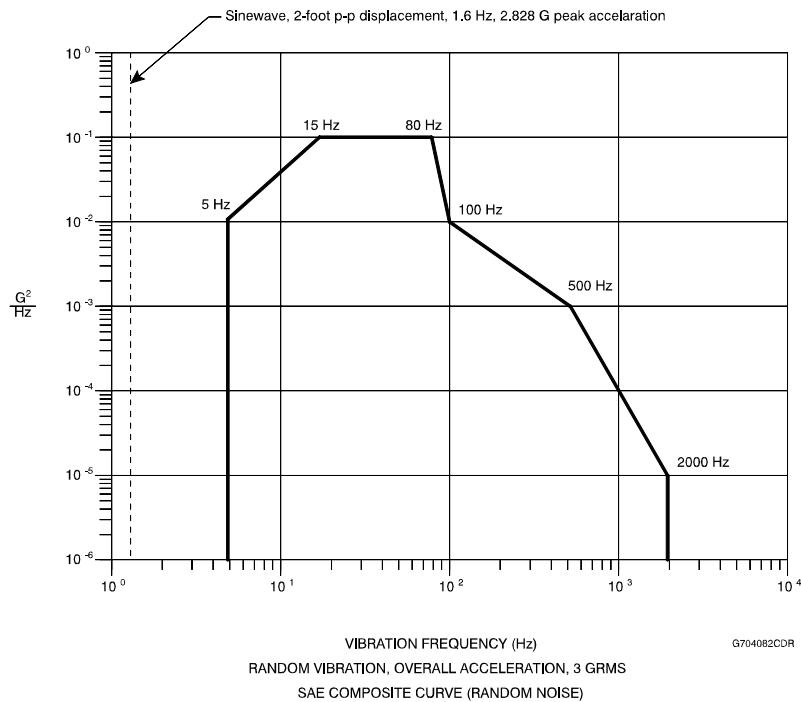


Figure 2-5. SAE Composite Curve (Random Vibration)

## DESIGN AND CONSTRUCTION

### A. Materials, Processes and Parts

The selection of parts and materials is based on commercial parts suitable for automotive and airborne applications. Standard parts and materials are procured to supplier's catalog number. All parts and materials are subject to CMC incoming inspection for conformance to requirements. Non-standard parts are also subject to CMC incoming inspection and documented on a CMC Source Control Drawing which include as a minimum the following:

- Electrical and mechanical characteristics
- Environmental and Quality Assurance requirements
- Workmanship requirements
- Marking requirements

Manufacturing processes used are selected for their full compliance to airborne requirements and are under statistical process control. All manufacturing processes are fully documented.

### B. Equipment Markings

The receiver part number shall be 220-600944-40X. The last digit shall be used to identify specific variations.

#### 1. S/W Part Number

The S/W part number appears on a label on top of the FEPRM. The host shall get access to the latest S/W part number on the serial interface port.

#### 2. Modification

A modification record numbered from 1 through 5 shall be etched on the PCB close to the H/W part number. This modification number shall be used primarily for H/W changes.

#### 3. Electrostatic Discharge Protection

The receiver shall be identified with a "Caution" Label. The receiver can withstand an electrostatic discharge level of 2kV from 100pF through 1.5kΩ between any two pins in either polarity (Mil. Std.883 human body model).

### C. Built-In Test (BIT) Requirements

The receiver performs self-tests and generates status information to provide an indication of the operational readiness and facilitate maintenance actions.

Failure indication is transmitted on the primary serial output bus via the self-test result (message ID #51). 90% of all receiver failure modes are detected and annunciated or have no effect on receiver outputs.

#### D. Interchangeability

Interchangeability of the receiver with any other receiver bearing the same part number shall not necessitate readjustments of any component in order to meet the performance requirements.

### HARDWARE INTERFACE

This section applies to the OEM board version of the receiver.

For details on the hardware interface of the Development Kit version of the receiver, please refer to Appendix A.

For details on the hardware interface of the STARBOX version of the receiver, please refer to Appendix B.

#### A. Connectors and Connector Pins Assignment

##### 1. Pin Assignment

Refer to Appendix C for the pin assignment.

##### 2. General

The receiver has two standard connectors. J1 is a 26 pin connector for general input/output interfaces and power input and J2 MCX type RF connector.

VAR -401      1mm Flexible Printed Circuit, 26 pin ZIF connector J1

The receiver is also available in different variations:

VAR-402      with a 0.100 x 0.100, 20 pin (2x10) Right Angle Shrouded Header with detent windows J3 instead of the ZIF connector J1.

VAR -403      with a 0.100 x 0.100, 20 pin (2x10) Straight Header J3 instead of the ZIF connector J1 on the TOP side.

VAR -400      with a 0.100 x 0.100, 20 pin (2x10) Straight Header J3 instead of the ZIF connector J1 on the BOTTOM side.

See Appendix C for more details.

### 3. J1 Interface and Power Connector

The J1 Interface Connector is a 1mm Flexible Printed Circuit, 26 pin, AMP 2-487952-6 or ELCO 00-6200-026-032-800. Following is the list of possible mating Flat Flexible Cable and connector manufacturers:

- a) AXON' CABLE (cable)  
390 E. HIGGINS Road  
ELK GROVE VILLAGE, IL 6000  
TEL: (708) - 806 - 6629
- b) MIRACO (mating and connector)  
9 PITTSBURG Av.  
P.O. BOX 1163  
NASHUA, NH 03061-1163  
TEL: (603) - 882 - 6887
- c) ELCO USA (connector)  
3250 KELLER Street, Unit One  
SANTA CLARA, CA 95054  
TEL: (408) - 499 - 1861

### 4. J2 RF Input Connect

The J2 RF input connector is an MCX Sub-miniature Snap-On Connector straight jack receptacle. The following is the list of possible mating connectors compatible with RG316 cable type:

Right angle: OMNI SPECTRA, 5807-5001-09 or SUHNER, 16 MCX-50-2-5C/111 or RADIALL, R113182.

Straight: OMNI SPECTRA, 5831-5001-10 or SUHNER, 11MCX-50-2-10C or RADIALL, R113082.

OMNI SPECTRA (M/A COM)  
100 Chelmsford St.  
P.O. Box 3295  
Lowell, MA 01853-9910  
TEL : 1-800-366-2266

HUBER & SUHNER  
One Allen Martin Drive  
P.O. Box 400  
Essex, VT 05451  
TEL : 1-802-878-0555

RADIALL  
150 Long Beach Blvd.  
Stratford, CT 06497  
TEL : 1-203-386-1030

## 5. J3 Interface and Power Connector

The J3 Interface and Power connector is a 0.100 x 0.100, 20 Pin Header (3 examples of manufacturer's part numbers):

AMP	1-103783-0
BERG	67996-120
SAMTEC	TSW-1-10-07-S-D

or a 0.100 x 0.100, 20 pin Right Angle Shrouded Header with detent windows (1 example of manufacturer's part number : connector and mating):

Connector:	AMP	102570-8
Mating:	AMP	87835-4

NOTE: Internal row contains the odd pin number (1-19)  
External row contains the even pin number (2-20)

## B. Power Input

The receiver shall operate from regulated DC power supplies as specified in Figure 2-6.

PIN NO.	FUNCTION (NOTE 1)	VOLTAGE	STANDBY CURRENT (TYP) mA NOTE 4	ACTIVE CURRENT (TYP) mA	ACTIVE CURRENT (MAX) mA	RIPPLE MAX. (NOTE 2)
J1-21	+5V Digital	5V +10%/-5%	18	70	130	100 mV
J1-26	+5V RF (Note 1)	5V $\pm$ 5%	15	75	110	50 mV
J1-15	VDD (Note 3)	5V +10%/ -5%	0.180	90	170	100 mV
	VDD (Note 5)	2.6V	0.030			

Note:

- To avoid CMOS latch-up condition, the maximum  $\Delta V$  (including ripple) between the +5V Digital, +5V RF and VDD shall be <0.5 V.
- Ripple specification is defined for frequencies up to 100 kHz.

Figure 2-6. Power Input (Sheet 1 of 2)



3. If the application doesn't request the SRAM Keep-Alive Mode (see para 3.1.1), this pin must be connected to J1-21 (+5V Digital).

Typically, the data will stay valid for VDD down to 2.6V but it is not guaranteed for all variations when  $VDD < 4.5\text{Volts}$ . Only for variations (contact CMC for more information) having Low voltage data retention SRAM, the data will be kept valid down to 2.6Volts.

The time source will be kept valid for VDD down to 2.6 Volts for any variations.

4. The Standby Current is measured when the Power Control Input is LO or when the +5V Digital is below the 4.5V threshold.
5. VDD current in SRAM Keep Alive Mode.

Figure 2-6. Power Input (Sheet 2 of 2)

#### 1. Power Control Input

The receiver possesses its own circuitry to perform a proper power-down and power-up sequence in order to preserve the non-volatile data in SRAM. The Power Control input allows also the possibility to generate a master reset (Standby Mode) to the receiver without removing the power.

A low voltage input will cause a master reset.

Refer to Appendix C for the electrical characteristics.

#### 2. Preamplifier Power Pass-Through (Antenna Supply)

The preamp signal is available on the I/O connector for the host to provide power to the antenna preamplifier via the centre conductor of the RF cable J2. The receiver is capable of handling voltages in the range of +5V to + 16V.

Note: Maximum current is 100 mA on J2.

#### 3. RF Input

The receiver will receive the GPS signal from the antenna amplifier on one RF input connector, J2. The RF input port impedance is 50 Ohms nominal with a maximum return loss of -10 dB over the frequency range of  $1575.42 \pm 3 \text{ MHz}$ . The nominal source impedance presented by the antenna shall be 50 Ohms with a maximum return loss of -10 dB.

#### 4. Discrete Inputs

For normal operation, all discrete inputs can be left opened. See Appendix C for the electrical characteristics.

## a. DISC\_IP\_1

The discrete input Discrete #1 is used to control the reprogramming of the Operational software. Refer to Appendix F for details on the programming mode procedure.

## b. DISC\_IP\_2 , DISC\_IP\_3 and DISC\_IO\_1

These 2 discrete inputs and the configurable discrete I/O signal are general purpose default condition inputs. (DISC\_IO\_1 can be configured as a discrete output signal for custom applications).

Note: Not all signals are available depending on the type of connector selected.  
(see Appendix C)

**TIME MARK OUTPUT 1 PPS**

The Time Mark discrete output interface is implemented using a standard TTL Logic output type. Clamping diodes are provided to Vcc and Ground, and the output is current limited using a series resistor. The time mark is a 1 Hz signal with its rising edge corresponding to the time when the navigation outputs are valid. (see Figure 2-6 for Time Mark waveform).

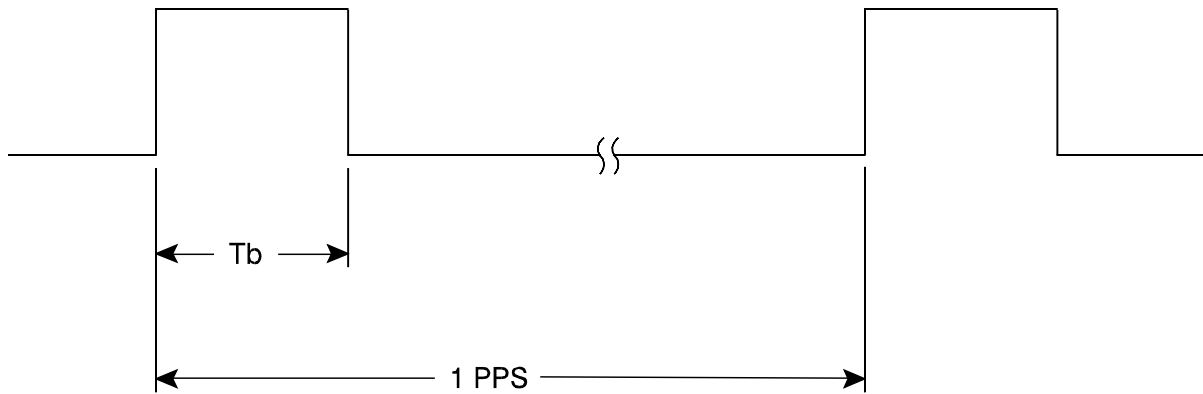
The Time Mark Output has 2 operating modes: Aligned on GPS Time or Free-Running.

In Aligned on GPS Time mode, the Time Mark Output and GPS measurements will be aligned on GPS time at  $\pm 200\text{ns}$  typically. With respect to Figure 2-7,  $T_b$  is  $1.01\text{ s} \pm 0.01\text{ ms}$ . To allow the synchronization on GPS Time, a maximum delay of 5 seconds can be added to the TTFF. See CMC Binary message ID #20 and #103 for more information.

In Free-Running mode the Time Mark won't be aligned and the TTFF is according to the specification. With respect to Figure 2-6,  $T_b$  is  $1.01\text{ ms} \pm 0.01\text{ ms}$  and occurs once each second approximately ( $999.999\text{ms} \pm$  receiver clock drift) with the rising edge (0 to 1 transition) corresponding to the receiver epoch (1 Hz).

In 2 Hz PVT mode, the Time Mark will be output once per second. In Time Alignment mode, the Time Mark will be synchronized to the Seconds boundary of the GPS Time.

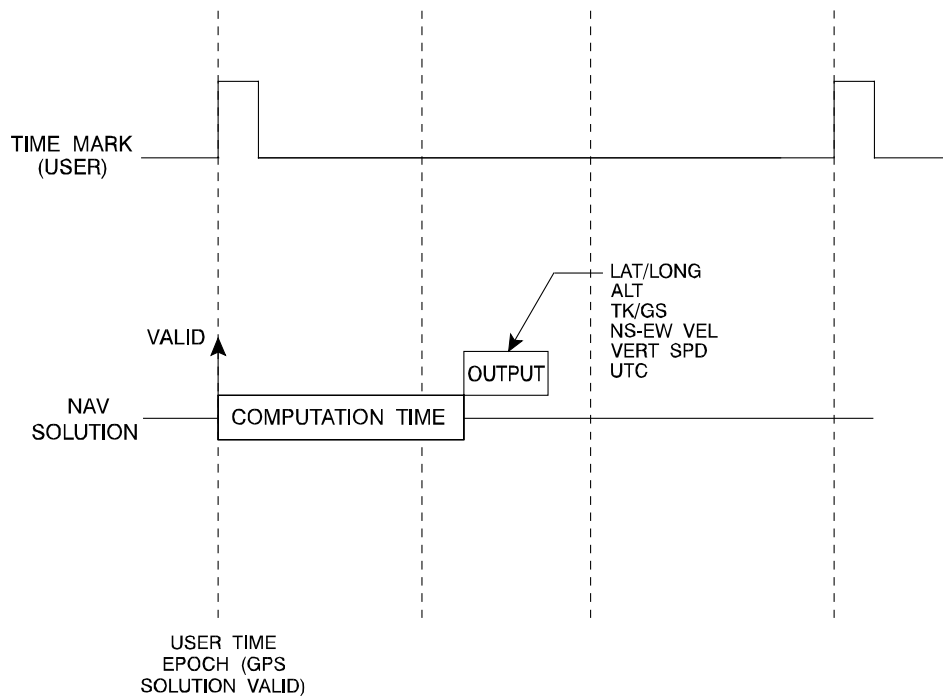
The Time Mark Output can also be configured as a standard discrete output fully controlled by the software for customized versions. See Appendix C for the electrical characteristics.



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Figure 2-7. Time Mark Waveform

The timing relationship for the GPS Time Mark output from the receiver is defined in Figure 2-8. The Navigation Data message ID #20 defines the UTC time of the epoch. The rising edge of the Time Mark is accurate to within 1  $\mu\text{sec}$  of UTC.



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Figure 2-8. GPS Timing Relationships

## SERIAL DATA INTERFACE

The receiver includes 2 standard serial input/output interface ports and one optional port. The ports are designated the Primary Port and the Auxiliary Port. Both ports operate independently with baud rates adjustable from 300 to 38.4 K baud.

The Primary Port supports data input (for receiver configuration and control) and data output (navigation results, receiver status etc.).

The Auxiliary Port supports data input (roving unit mode) or output (base station mode) for differential correction data adhering to Ref [2].

Both ports can be used for S/W reprogramming (refer to Appendix F).  
Refer to Appendix C for the electrical characteristics.

### A. Primary Port

The Primary Port supports communication via the CMC Binary protocol. Through specific CMC Binary messages, the primary port is re-configurable to communicate with a PC-based Monitor named StarView (for extensive monitoring of SV tracking, measurements and navigation status).

The default baud rate is 9600 but can be reconfigured (see CMC Binary message ID #110). If no default message list has been stored in NVM, the receiver will output the CMC Binary message ID #20 at a rate of once per second after each power up.

### B. Auxiliary Port

The auxiliary port input is used to receive (roving unit mode) or transmit (base station mode) RTCM differential messages (Ref [2]). The default baud rate is 9600 and can be modified via the CMC Binary Set DGPS Configuration message ID #83.

The new configuration will be stored in NVM.

The output port is used to transmit RTCM differential message when the receiver is acting as a base station.

## NON-VOLATILE MEMORY DATA

The receiver stores in NVM different types of information used to accelerate the TTFF and to configure the I/O; refer to Figure 2-9 for a partial list of data stored in NVM.

PARAMETER	NOTES
ALMANAC	The most recent one
LAST POSITION	Position in NVM is updated at different rates depending on the application. The last known position is always kept in battery back-up SRAM.
DGPS CONFIGURATION	
RS232 CONFIGURATION	Contains the following configuration information :  1. Mode of operation 2. Baud Rate: 300 to 38400 3. Default CMC Binary message list 4. Time Alignment Mode State 5. Mask Angle 6. Used Datum
BASE STATION PARAMETERS	Position and message rates (base station configuration only)

Figure 2-9. Non-Volatile Memory Data

## SECTION III - INSTALLATION AND VERIFICATION

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## SECTION III

### INSTALLATION AND VERIFICATION

This section covers the installation and verification of the receiver. Sold separately, the StarView Graphical User Interface running on a PC allows the user to control the receiver and to display its outputs. Details on StarView are provided in Ref. [5].

#### EQUIPMENT REQUIRED

Refer to Figure A-1 in Appendix A for a description of the equipment required for the receiver to operate.

#### ELECTROSTATIC DISCHARGE WARINESS

Refer to the electrostatic discharge notice in the preliminary pages of this manual.

#### EQUIPMENT INTERCONNECTION

As aforementioned, the receiver can be provided either as an OEM board, within a STARBOX unit or within a Development Kit. The interconnection of the OEM board format is guided by its physical and electrical specifications detailed in the previous section. A complete description of the Development Kit is provided in Appendix A and a complete description of the STARBOX is provided in Appendix B.

The receiver includes two serial communication ports: COM1 and COM2. Serial communication with the receiver must be performed on COM1. The I/O protocol is discussed in a subsequent section. The minimal baud rate is 19200. The other serial port, COM2, is used for the differential link, and its minimal baud rate stands at 9600.

#### ROVING UNIT/BASE STATION CONFIGURATION

Athe receiver may be configured either as a roving unit or a base station via the command message #80. Reception of this command will revert to the requested mode on the fly. Note that no effect will take place if the system is already in the requested mode.

Each receiver is shipped configured as a roving unit by default. The user can then choose which unit(s) will be the base station(s). A base station may then be swapped with another receiver if for some reason it is damaged. This will permit the user to continue operating his system, provided sufficient receiver units are available.



## INSTALLATION CONSIDERATIONS

All formats of the receiver are not waterproof, therefore they must be mounted in a dry location. They should also be located where it's convenient for cables to run to the power source, display device, and antenna. Drip loops should also be formed to prevent moisture from running down the cables and into the receiver.

The receiver should be mounted several feet away from radio transmission equipment.

### A. ANTENNA LOCATION

1. The antenna must be mounted high enough to provide an unobstructed view of the sky in all directions. The receiver uses satellites that can be as low as 5° above the horizon, so nothing should block it from the sky. Ensure that the bottom of the antenna is at least 5 inches above the surface it's mounted on. The antenna should also be mounted below the radiation plane of INMARSAT or radar antennas, and away from any other high-power transmitting antennas.
2. Care should be taken as well to avoid coiling the antenna cable around the mounting base and pinching the antenna cable in window or door jambs.

### B. BASE STATION LOCATION

1. It is imperative that the base station be located on a site that is above all obscuring elements on the surrounding terrain in order to have all satellites above the horizon visible at the base station's antenna. The intent is to have all satellites that are visible at the roving user's antenna to be visible at the base station as well.
2. As well, multipath interference must be minimized as much as possible. Multipath is defined as the interaction of the GPS satellite signal and its reflections; this causes errors mainly on the GPS code, but not so much on the GPS carrier. Even though the receiver uses carrier phase measurements, it can revert to code differential GPS operation if carrier phase differential GPS cannot be performed. Hence, the base station's antenna must be far from any reflecting elements.
3. The position of the base station's antenna must be surveyed using appropriate surveying equipment. This position must then be programmed in the base station using the message ID #80. Any error in the base station's position will be reflected in the roving user's computed position.

### C. DATA LINK

1. The data link must operate at a minimal rate of 9600 bauds.
2. The required power level depends on the distance separating the base station and the roving units.

### D. BASE STATION AND ROVING UNITS SEPARATION

The operational range of carrier-phase differential measurements is limited to about 20 km, after which significant accuracy degradation could occur. If your application requires greater separations, your own base station network must be established.

## CHOICE OF A WIRELESS DGPS DATA LINK

The choice of wireless link is a key part of any DGPS system. The functionality and reliability of the link can have a significant effect on the success of the DGPS system. The key functional parameters affecting the performance and cost of DGPS wireless links are:

- Range
- One-way versus two-way data capability
- Latency and rate of data transmission
- Radio Frequency
- Frequency Selector
- Interference Rejection
- Wide area Differential network capability

For narrow-band communication, typical frequencies of operation are in the 150 MHz or 450 MHz bands. Data rates range from 9600 to 19.2 kbps. RF transmit power ranges from 2 W to 30 W.

For spread-spectrum links, 900 MHz or 2.4 GHz is typical. Data rates range from 19.2 to 115 kbps. Power is 1 W or less. These are typically short range (<5mi.) links for portable or mobile operation.

### A. RANGE

Exact range of a wireless radio link is difficult to calculate without a detailed engineering analysis. Reasonable approximations are possible however. Range is primarily affected by the combination of the following factors:

- Terrain
- Transmit power and receiver sensitivity
- Transmitter and receiver antenna gain.

The simplest calculation of range assumes the earth is smooth and spherical. This is the starting point for all range calculations and establishes the minimum height requirements for the antennas. The calculation establishes range by line of sight. The range in miles is given by.

$$\text{Range} = \sqrt{(2 * H_t)} + \sqrt{(2 * H_r)}$$

where  $H_t$  is the height of the transmit antenna in feet and  $H_r$  is the height of the receive antenna in feet. Given 8 feet height for the receive antenna and 25 feet height for the transmit antenna, the range is 11 miles. Note that if the transmit antenna or receive antenna are on hilltops, the height of the hill above the highest terrain between TX and RX should be included in the height of the antenna.

### Terrain

Terrain is the greatest contributor to short range (<100 miles) communication. Terrain includes the shadowing or blocking effect of hills and valleys as well as buildings and foliage. Dense foliage can easily shorten a smooth earth range calculation of 10 miles to 2 or 3 miles. Foliage can often be overcome by brute RF transmit power or excellent receiver sensitivity.

## Transmit Power and Receive Sensitivity

Transmit power and receiver sensitivity can be traded off against each other in cases where you are not competing with another user on the same frequency. Having a receiver sensitivity of say 6 dB better than a competing receiver makes your transmitter look 6 dB (4x) more powerful. This translates to more range and a more reliable link. Alternately, having a sensitive receiver can significantly lower the cost of the transmitter by allowing a lower power model. Having a lower power transmitter can increase battery life or reduce battery weight in portable applications.

## Antenna Gain

Antenna gain increases the effective radiated power of a transmitter and the effective sensitivity of a receiver. A 5 W transmitter with a 6 dB (4x dipole) gain antenna looks like a 20 W transmitter when compared to the same unit on a simple vertical whip with a ground plane. A receiver with a 6 dB antenna sees a 5 W transmitter as if it has raised its power to 20W. With a 6 dB antenna on both transmit and receive, the 5 W transmitter performs like an 80 W transmitter in the case of 0 dB antennas on both ends.

### B. ONE WAY VS. TWO WAY LINK

In many applications such as DGPS, it is only important to send a message one way. In this case the wireless link can be made less expensive by using transmit only and receive only radios. This can also reduce the cost, size and weight of the link. Two way is useful in applications such as tracking, AVL and dispatch where the data must be sent back to the base.

In two way applications that have high update rates and/or a large number of users, key performance items to look for are over-the-air data rate and data turnaround time. See Latency and Rate of Data Transmission below.

### C. LATENCY AND RATE OF DATA TRANSMISSION

Latency and rate of data transmission can have a significant effect on the number of users that can be supported on a single radio channel as well as the time it takes to get an update to the base. Latency is affected by the data rate at the serial ports of all the equipment in the link as well as the over the air data rate. The higher the data rates, the lower the latency or age of DGPS corrections. The higher the data rate, the higher the number of updates or DGPS corrections per second.

### D. THE RADIO FREQUENCY USED

The radio frequency can have some effect on the link results. Low frequencies tend to propagate better over terrain and higher frequencies tend to be more line of sight. For a given amount of antenna gain, higher frequency antennas are smaller in direct proportion to the frequency difference. The higher gain antennas also tend to be less expensive at higher frequencies due to their smaller size.

It should be noted that at the higher frequencies (above 400 MHz), transmission line loss must be considered. A run of 50 feet using an inappropriate cable can easily lead to a loss of half of your transmit power or more. The same applies to the receive side of the link in terms of loss of effective receiver sensitivity.

## E. FREQUENCY SELECTOR

Many DGPS links in North America are operated on a small group of itinerant frequencies. These frequencies can become congested in urban areas. Most radios are synthesized and can be programmed to operate at a specific frequency or set of frequencies if equipped with a selector switch. Having the selector switch under field conditions can greatly simplify changing frequencies in the case of interference from other users on a frequency. Having a larger number of positions on the selector switch can give a greater choice of alternate frequencies.

## F. INTERFERENCE REJECTION

Common forms of interference are:

- Co-channel
- Image channel
- Intermodulation
- Adjacent channel.

Co-channel interference is when someone is operating on the same channel as your wireless link. The simplest ways to eliminate this are to relocate to a different channel or to use more power than the competitor. Note that using more power means that your receiver must see you base station at a higher power level than the competing station.

Image channels are channels that are separated from your channel by 2x the first intermediate frequency (I.F.) of your receiver. A common I.F. is 21.4 MHz. With poor image rejection, a channel that is 42.8 MHz away from your channel can strongly interfere with your desired signal. An external preselector can minimize this problem. Some radios are available with high selectivity preselectors already built in and thus minimize the tangle of extra cables and bulk of the external unit.

Intermodulation (IM) interference is a complex process where two channels mix to generate a signal that is on your channel. This mixing can take place in the DGPS wireless link receiver. Some types of IM can be reduced by having a good preselector on the receiver front end thereby attenuating one or both of the offending signals. Close in frequencies simply require a good IM performance specification. Look for an IM specification in excess of 60 dB.

Adjacent channel interference typically occurs when there is a strong signal in the next adjacent channel and you are near the limit of range of your system. Look for specifications in excess of 65 dB.

## G. NETWORK CAPABILITY

In some cases, a DGPS reference station with single transmitter cannot cover enough area without the logistical difficulty of frequently moving the station. A wider area can be covered using a single reference station with multiple transmitters. The Network uses the first transmitter to send the DGPS correction and it is in turn repeated by one or more distant transmitters.

To set up a DGPS network with several repeaters requires the wireless link to have a network protocol capability. Protocols such as AX.25 or the more powerful MX.25 support powerful features such as multi hop digipeting (digital repeating) and time slotted digipeting. Systems have been set up that cover more than 30,000 square miles using a single DGPS reference station. Systems can even include mobile, marine or airborne repeaters without a degradation of DGPS accuracy.

#### H. WIRELESS DGPS LINK OPTIONS

CMC is teamed with GLB for many of their DGPS link products. They have been used extensively in the field with our products and have proven to be reliable and efficient. GLB offers wireless links that have been extensively used for DGPS applications. Features include:

- Multiple channel selector switch.
- High receiver sensitivity.
- Built in preselector for image interference rejection.
- Powerful AX.25 and MX.25 protocol for repeating or network coverage.
- High speed 9600 bps operation.
- Fast turnaround time for Tracking and AVL.
- Rugged water resistant packaging.
- 5 W and 25 W transmitters.

#### GLB Products

All configurations are available with single or multiple frequency selector. Complete kits are available with Antennas, and RF/data cabling. 150 MHz and 450 MHz frequency bands are available from stock. Other frequencies in the 125 MHz to 960 MHz band are available on request.

Typical 450 MHz configurations are:

450 MHz TX only 5 W	SN2TX96-450-5
450 MHz RX only	SN2RX96-450
450 MHz 25 W TX only	SN2TR96-450-25
450 MHz TX only 25 W Ruggedized Enclosure	BASE8-450-25
450 MHz TX/RX 5 W	SN2TR-450-5

Please contact GLB for Plug and Play Packages that include antennas and RF/data cables.

GLB Electronics Sales 905-878-7794  
<http://www.glb.com>

Technical 716-675-6740

## SECTION IV - OPERATION

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## SECTION IV

### OPERATION

#### RECEIVER STATES

##### A. NON-OPERATIONAL STATES

The receiver has two non-operating modes, OFF mode and SRAM Keep-Alive mode. The maintenance of the DC power on the SRAM will determine which of the two non-operating modes will be automatically entered during the power down sequence.

###### 1. OFF Mode

In OFF mode only the data contained in the NVM is retained for use when power is re-applied. Refer to the NVM Data section details on retained data.

###### 2. SRAM Keep-Alive Mode

In SRAM Keep-Alive mode specific data contained in the SRAM is retained to reduce the time-to-first-fix when power is re-applied. Data retained in SRAM mainly consists of valid satellite ephemeris data not older than 3 hours.

##### B. OPERATIONAL STATES

1. The receiver has 6 operating modes: Self-Test, Initialization, Acquisition, Navigation, Dead-Reckoning and Fault. The receiver switches between modes automatically as shown in Figure 4-1. The receiver reports on its host port the current operating and navigation modes.

###### 2. Self-Test Mode

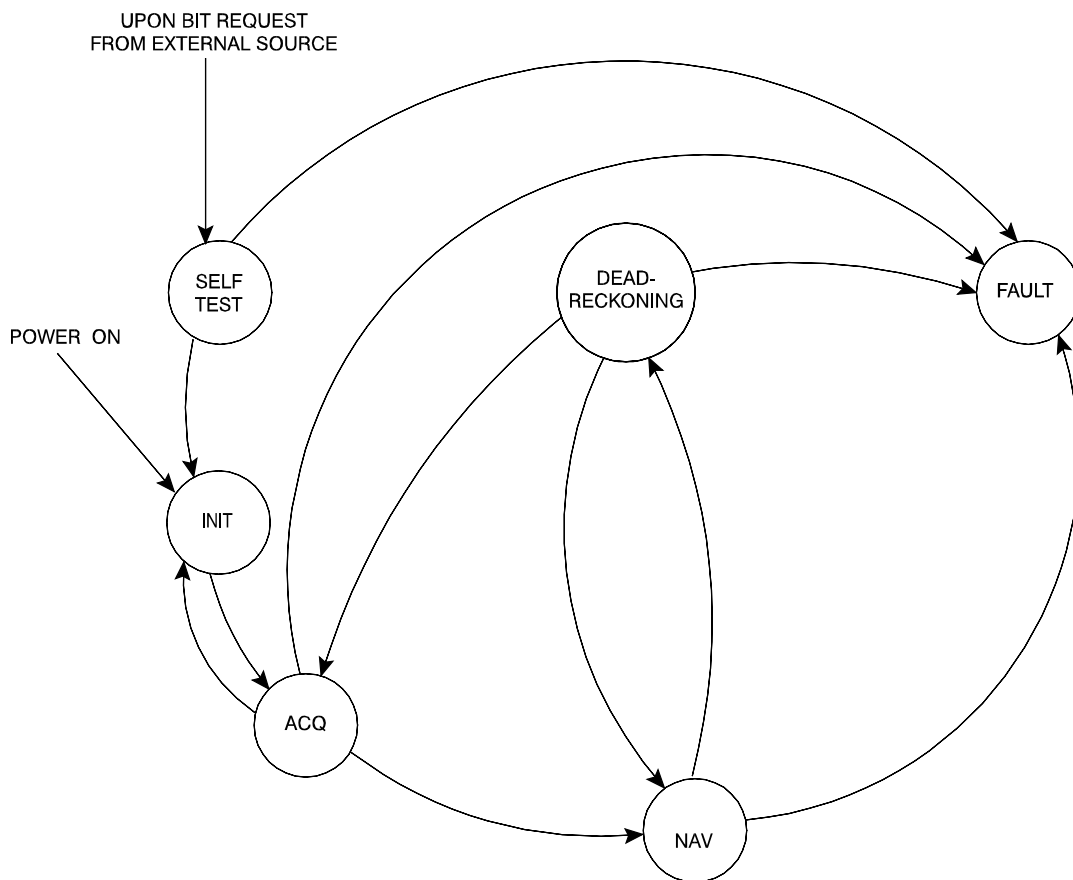
The receiver enters Self-Test mode upon request from an external source (CMC Binary message ID #51). The time duration spent in the Self-Test mode is no more than 15 seconds. On self-test completion, the receiver reports the BIT results on its host port through the CMC Binary message ID #51. Self-Test mode exits to either Initialization or Fault mode.

###### 3. Initialization Mode

Upon power-up, the receiver enters Initialization mode. During this mode hardware is initialized prior to Acquisition mode entry. The Initialization mode is also initiated upon completion of the Self-Test mode, but exits always to the Acquisition mode.

Depending on the previous non-operating state (OFF or SRAM Keep Alive Mode) the receiver will retrieve data only from the NVM (cold start) or from both NVM and the SRAM (warm start). Integrity checking is done on all data retrieved from the non-operating state.





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Figure 4-1. Receiver Operating Modes

During initialization, the receiver retrieves the last received valid almanac data and last user position from NVM, gets the current time from the low-power time source and predicts which satellites are currently visible. This list of visible satellites is then used in Acquisition mode to program the 12 parallel correlator channels.

#### 4. Acquisition Mode

The receiver is in Acquisition mode when insufficient satellite data is available to produce an initial navigation solution. Acquisition mode is entered from Initialization, or Dead-Reckoning mode, and exits to Navigation or Fault mode.

To acquire signals from the GPS satellites, the receiver uses:

- a. Almanac data which describes the satellite orbits.

- b. Time, which in conjunction with almanac data is used to estimate the present position of satellites in their orbits.
- c. The approximate location of the receiver so a prediction can be made as to which satellites are visible.

The receiver then collects ephemeris data by decoding the satellite down-link data message. After each satellite in view is acquired, its measurement data set is produced. When a sufficient number of satellites are being tracked, position, velocity and time can be computed and Navigation mode entered.

If the receiver cannot perform an acquisition due to an absence of valid almanac data or user position and/or time it initiates a "Search the Sky" acquisition. The receiver attempts to acquire all satellites in the GPS constellation. Once a satellite has been acquired, ephemeris data is decoded from the satellite down-link message. After sufficient satellites have been acquired, the receiver enters Navigation mode. In "Search the Sky", the TTFF is typically less than 3 minutes.

## 5. Navigation Mode

The receiver is in Navigation mode whenever sufficient satellite information and measurement data is available to produce a GPS fix. Navigation Mode is entered from Acquisition or Dead-Reckoning mode, and exits to Dead-Reckoning or Fault mode.

In Navigation mode, a receiver configured as a roving unit operates in 2 sub-modes: Differential and Stand-Alone Nav. Sub-mode transition occurs automatically depending on satellite data availability. A receiver which is configured as a base station unit will operate in Base Station Navigation mode only. The receiver reports its current navigation sub-mode on its host port.

### a. Differential (Roving Unit Only)

The receiver operates in Differential mode when data from at least 4 satellites with adequate geometry and differential corrections and/or measurements exists to compute position, velocity and time outputs. This is the preferred navigation mode. Differential data is supplied to the receiver via the differential input port. Differential data can be received only on the auxiliary serial data port.

### b. Stand-Alone Nav (Roving Unit Only)

The receiver operates in Stand-Alone Nav mode when data from at least 4 satellites with adequate geometry, but no differential corrections or measurements, exists to compute position, velocity and time outputs. This is the preferred navigation mode when insufficient differential data is available to generate a differential GPS fix.

c. Base Station Nav (Base Station Unit Only)

The receiver operates in Base Station Nav mode once the time has been initialized and at least 4 satellites with adequate geometry can be used for navigation purposes. Once in this mode, only a change of configuration (rover mode requested) or a reset will cause the unit to leave this navigation mode. In this mode, the unit will have the ability to transmit the DGPS messages which are requested and allowed once its position is initialized. (Refer to the Configurable Parameters section for position initialization details.)

7. Dead-Reckoning Mode

The receiver enters Dead-Reckoning mode when it cannot remain in a Navigation. The speed and direction is assumed constant to allow the receiver to provide an estimated position.

8. Fault Mode

The receiver enters Fault mode during the period of the time in which the receiver outputs are affected by one or more critical system faults. This mode supersedes all others and remains active until the next power-down/power-up cycle. Fault mode is entered from any other mode except Initialization.

C. DATUM SUPPORT

The receiver has the ability to provide its position in one of the 62 predefined datums. The list of all the supported datum is provided in Appendix E. Moreover, the receiver can also support two user-defined datum. These have to be defined, prior to their use, using binary message ID #88. Afterwards the desired datum, whether it is user-defined or predefined, can be selected using CMC Binary message ID #88.

## POWER-UP INFORMATION

At power up, the receiver sends two categories of factory information data to the main port (COM1) at 9600 bauds. The categories of information, Boot and Operational information, can be displayed on a dummy terminal.

A. BOOT INFORMATION

The Boot information contains the following factory data:

ALLSTAR

V4

G : XXXXXXXXXX

**169-613914-007** : Boot S/W Part Number

D0

PCPB: XXXXXXXXXX

GO : Go in Operational Mode

## B. Operational Information

The Operational information contains both the factory and the current operating mode information. The current operating mode baud rate is output twice. This is useful when the operating baud rate is not 9600.

Example :

```
<CMC,Part Nb:169-614110-XXX1, CB=0x0000003F2 SHP  
Go to CMC Binary @ 19200 baud.  
In CMC Binary @ 19200 baud3.  
|>3
```

Note 1: Operational S/W Part Number

Note 2: Power-up BIT result.

Note 3: Line transmitted at the Configured Baud Rate

## CONFIGURABLE PARAMETERS

Several parameters of the receiver and the base station are configurable and therefore, must be defined by the user prior to operation.

### A. BASE STATION CONFIGURATION

The auxiliary communication port COM2 is used to output the differential data. Its baud rate is fixed at 9600 bauds. The base station must be properly configured in order to encode the differential messages. The two elements that are required are: a precise position in order to be able to calculate the corrections, and the list of messages that are to be transmitted by the base station with their transmission rates. Two CMC binary messages permit the configuration of these elements through the main communication port COM1. These messages are: Set Operating Mode (message ID #80 ) and Set Differential Message Rate (message ID #91). Once these parameters have been set once on a unit, they will be kept in NVM. This prevents having to reconfigure a base station unit every time the power is cycled.

Since a precise knowledge of the base station position is required before being able to operate, its exact position must be saved in NVM. The base station is capable of self-surveying its position. The precision obtained from a self-survey operation is in the order of 2 meters. For a better precision, the position of the antenna must be surveyed using precise surveying equipment.

#### 1. Configuration Messages

Message ID #80 permits the following :

- set the base station's ID (number from 0 to 1023)
- set the base station's health (refer to Ref [2] for health codes)
- set the base station's position with a known position or the current self-surveyed position
- start a Self-Survey operation for a defined time period

#### Self-Survey operation:

Upon entry in the base station navigation mode, the self-survey process is started and will continue as long as the unit stays in the base station navigation mode. There are two ways to take advantage of the Self-Survey capability: the Set Survey command and the Get Survey command which are both supported by message ID #80.

- Set Survey resets the current self-surveyed position and restarts the self-survey process for the desired period of time (maximum duration: 48 hours). At the end of the period, the computed position will automatically be used as the base station's position and the encoding of the differential messages will start if the differential message rates have been configured properly.
- Get Survey will take a snapshot of the current self-surveyed position and save the base station position with this calculated position. It will start the encoding of the differential messages if the differential message rates have been configured properly.

#### NOTE:

1. A Get Survey request will automatically stop any survey request initiated by a Set Survey request.
2. A Get Survey request can be performed at any given time. It configures the base station position with the position computed by the self-survey process.

Message ID #91 sets the following parameters:

- the differential message type (i.e. the RTCM message types)
- the differential message transmission period (0 to 255, where 0 stops transmission)

The detailed contents of these messages are provided in the Serial Data Interface section.

#### 2. Moving a base station

Special care must be taken when moving a base station that has been configured with a position. Since the base station configuration is saved in NVM, this configuration must be invalidated when the power is reapplied on the base station if roving units are monitoring the base station. It is recommended to set all the differential messages rate to 0 before moving the base station. This will avoid the roving stations from generating erroneous solutions.

#### 3. Configuration process example

Here are the typical steps that should be followed in order to achieve an easy base station installation:

- a. Power-on the unit .
- b. Using message ID #80, send a Set Self-Survey request to the unit (recommended survey time of 10 minutes). This will put the unit in base station navigation mode.
- c. Using message ID #91, set the desired differential message rates. At this point, the base station should start to generate corrections.
- d. Power-up a roving unit and establish the radio link between the two.
- e. Once the radio link has been established and everything works fine, the base station should now enter a longer self-surveying period to ensure sufficient accuracy of the base station's position. Using message ID #80, send a Set Self-Survey request to the unit (recommended survey time of 24 h). The base station will automatically start to transmit the corrections at the end of the survey period.
- f. Reset the roving unit. This will permit to start navigating with knowledge of the proper base station position when the survey process is finished.

**Note:**

If the precise base station position is already known, set the base position using message ID #80 in step b. and skip steps e. and f.

### C. MASK ANGLE

The mask angle is defined as the minimum satellite elevation angle (in degrees) above which any given satellite must be in order for it to be used in the GPS position solution. Low satellites usually do not yield accurate measurements due to weak signal reception and possible multipath. Typical mask angle values range from 5°-10°, depending on the receiver's location. This value is programmable via command message #81.

### D. GPS ANTENNA POSITION

For the base station, it is imperative to program the surveyed position of the GPS antenna. This can be done using either the X-Y-Z coordinates in meters within the WGS-84 reference frame, or latitude and longitude in degrees as well as height in meters.

This can be achieved via message ID #80.

## DATA REQUESTS

Data may be requested for output by the receiver for display or logging purposes. The list of data request commands and data messages is detailed in the following section.

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## SECTION V - SERIAL DATA INTERFACE

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## SECTION V

### SERIAL DATA INTERFACE

#### SERIAL DATA PROTOCOL

The purpose of this section is to define a serial data transfer protocol for the receiver. The serial data is transmitted in variable size message blocks, where the message block header defines the contents and size of all message blocks bearing this ID.

For discussion purpose, the transmitter is the controlling Host CPU, and the GPS receiver is the Receiver. Prior to entering the protocol, both the transmitter and receiver must be set up to the same baud rate and data setting. Upon entering the protocol, the transmitter and receiver wait for the possible transmission of message blocks.

#### A. PHYSICAL LINK LAYER

The electrical signals used are those for RS-232 communication port. Only the Receive and Transmit lines are required. The serial port is asynchronous and should be set up with 1 start bit, 8 data bits, no parity bit, and one stop bit. A default baud rate of 9600 is used. Both transmitter and receiver are operating at the same rate and can be reprogrammed (see msg ID #110).

#### B. DATA LINK LAYER

##### 1. Bit Ordering

The ordering of data within message blocks is such that the least significant bit (LSB) is the first bit received, moreover the most significant bit (MSB) is the last bit in the sequence.

	MSB				LSB			
Order	7	6	5	4	3	2	1	0

This ordering is applied to all data formats, which include integer values, fixed point values, floating point values, and character strings.

##### 2. Message Block Structure

All communication is done using message blocks. Each message block consists of a header and possibly data. The data portion of the block is of variable length depending on the message ID. The header has a fixed length of 4 bytes, consisting of a Start-of-Header character (SOH), Block ID, Block ID Complement and Message Data length. Each block has a truncated 16-bit word containing the Checksum associated with the complete content of the block. It is appended at the end of the Data portion of the block.

The Message Block structure is as follows:

byte 1 [SOH]  
byte 2 [ID #]  
byte 3 [Cmpl ID #]

byte 4 [Msg Data Length]  
byte 5 [Data Word 1] LSB  
byte 6 [Data Word 1] MSB  
.. ..  
[Checksum] LSB  
[Checksum] MSB

where:

SOH

Start of header character (decimal 1).

ID #

Byte containing the Block ID numeric value. The block ID number field is used uniquely to identify the format of the data portion of the block. Since only 7 bits are needed for the ID, the higher bit is used to encode information about start/stop of broadcast of data blocks and to set special modes for command messages. This prevents an unnecessary increase in overhead by eliminating any extra bytes in the protocol.

Cmpl ID #

1's complement of the ID # field. This can be calculated as  $\text{Cmpl Block \#} = 255 - (\text{Block \#})$  or using XOR as  $\text{Cmpl Block \#} = (\text{Block \#}) \text{ XOR } 255$ . This field, in conjunction with the Start-Of-Header, helps to synchronize the message blocks, since the SOH character can appear within the data, the Cmpl Block # field validates the header contents and thus confirms the start of the block.

Msg Data Length

One byte containing the length of the data part of the message in bytes (excluding header and checksum).

Checksum

This fields contains the checksum value for the complete message blocks transmitted, which includes header and data. The checksum calculations is discussed in more detail below.

### 3. Message Block Types

#### a. Host CPU to Receiver Message Types

There are 5 types of messages:

Dummy Message (ID #0):

Reserved

**Initiate Link (ID #63):**

First message (optional) to be sent by the transmitter upon entering the protocol. Its purpose is to inform the receiver that communication is desired. A password is encoded in the message. If the receiver was already transmitting data, this message will interrupt all output messages and will wait for new data request messages.

**Data Request Messages:**

Request the receiver to turn on/off the transmission of broadcast data or to transmit data only once. The MSB of the message ID will indicate the type of request with "1" to turn on broadcast, and "0" for once only or to turn off the broadcast.

**Command Messages:**

Request a particular receiver action other than a data request. The MSB of the msg ID may be used to set the receiver to normal mode (MSB=0) or to special mode (MSB=1).

**Data Messages:**

Any message containing data to be memorized or processed by the receiver.

**b. Receiver to Host CPU Message Types**

There are 6 types of messages:  
(All data is sent in receiver internal format)

**Dummy Message (ID #0):**

Reserved

**Initiate Link (ID #63):**

This is the response to the transmitter initiate link message.

**Acknowledge Message (ID #126):**

All transmitter messages are acknowledged by the acknowledge message. This message is sent as soon as possible if there is at least one message to acknowledge. The data field of this message contains 5 bytes which encode the IDs of the messages acknowledged (4 messages per time interval and possibly a message from previous time interval that was not completely decoded). So, a maximum of five messages may be acknowledged per message. ID #0 indicates a dummy message and should be discarded by the transmitter; its purpose is only to fill the data field of the acknowledge message block.

**Link Overload Error Message (ID #125):**

Sent by the receiver only when at least one output message caused an overload of the transmission link. This message is sent at a maximum rate of once per second. This message encodes a bit map of all the message IDs (#1 - #127), therefore indicating which IDs caused the link overload. The request of the message that caused the overload is cancelled to prevent any further overload.

**Data Messages:**

Messages containing requested data.

**Status Messages:**

Informs the transmitter on the status of a file transfer performed via a command message. The status is encoded in the MSB of the ID field. If the MSB = 0, the command request is unsuccessful. If the MSB = 1, the command is successfully performed. This message is sent within 1 minute after the command message. (This is currently only use for the almanac

**C. INITIATION**

Upon receipt of initiate link message block containing a valid password, the receiver sends a message block back to the transmitter with its own password.

This command also cancel all previous data request messages within 2 seconds.

The receiver will respond within 300msec to the initiate link command.

**D. DATA TRANSMISSION**

In most cases the receiver is given command message blocks for which it must respond with one or several blocks of data. Typically the following sequence of events occurs once the link is initiate.

The transmitter sends one or more message blocks to the receiver while keeping track of all message blocks that need to be acknowledged by the receiver. The receiver searches out each message block sent by the transmitter and then compare its own checksum calculation with the value that was sent by the transmitter. If the values match, the receiver includes that particular ID in the acknowledge message block. If the checksums are different, the receiver will not include the ID. Once all message blocks received during the last time interval scheduled by its executive are decoded a new acknowledge message block is built with all valid ID's received. The acknowledge message will be transmitted in the next available time slot.

For each individual message block transmitted, the transmitter must wait for its corresponding acknowledge or produce a time out error if not acknowledged within 300 ms.

The transmitter may send additional message blocks at any time. All message blocks are treated independently, therefore the transmitter do not need to wait for acknowledge before another message block can be transmitted, except for file transfer command messages, in which case the transmitter must wait for acknowledge message before continuing a file upload.

## E. ERROR RECOVERY AND TIMING

Error detection and recovery are incorporated in this protocol. Some of the common error conditions are listed below:

### 1. Block ID Complement Error

If the block ID in the header portion does not match the complement block ID number, the block must be discarded. This means that the data received is probably not a block.

### 2. Checksum Error

For the receiver, if the calculated checksum value on receipt of a block does not match the value in the block, the block must be discarded and this message block's ID is not indicated in the acknowledge message block sent to the transmitter. For the transmitter, if it detects a checksum error then the block must be discarded and a message block timeout should occur for the corresponding request.

### 3. Transmit Timeout Errors

The transmitter should wait up to the message rate for the reception of a data message block. Afterwards, the transmitter should report the error.

### 4. Frame Synchronization Errors

Since extraneous characters can be generated when using asynchronous communications, the receiver does not count on receiving valid blocks with no extra characters for each block transmitted. Synchronization is as follows: if the character received when expecting the start of a block is not a SOH, then it ignores the character and continues to search for a SOH. Once a SOH is found, the receiver assumes that the next two bytes are a valid block ID number and complement. If they are complements, then it assumes that the packet has begun and the search for the next SOH starts after the checksum even if the checksum is invalid. If they are not complements, it continues to search for SOH from the location of the block ID.

## F. CHECKSUM CALCULATION RULES

The 16-bit checksum is defined as the 16-bit sum of all the unsigned 8-bit bytes starting at the beginning of the header, any overflow or carry to the 16-bit sum is discarded immediately. Therefore, it adds unsigned bytes to produce a 16-bit result. For example, a valid initiate link message can be:

```
SOH, ID#, Compl ID#, Length, U, G, P, S, -, 0, 0, 0, Cksum(LSB), Cksum(MSB).
01, 63, 192, 08, 85, 71, 80, 83, 45, 48, 48, 48, 772 (decimal)
01h, 3Fh, C0h, 08h, 55h, 47h, 50h, 53h, 2Dh, 30h, 30h, 30h, 04h, 03h
(hexadecimal)
```

## G. DATA STRUCTURE

This paragraph describes the data representation standards to be used in formulating the contents of data fields. The structures defined are:

1. Character Data
2. Integer Values
3. Fixed Point Values
4. Floating Point Values

Character Data is to be stored in the following order in the Block data field:

5      . . . . . 8	7      . . . . . 0
CHAR 2	CHAR 1
CHAR 4	CHAR 3
CHAR 6	CHAR 5
CHAR 8	CHAR 7

Character Data are unsigned by default.

Integer Values are represented in two's complement form.

Floating Point Values are stored in IEEE format using "little-endian" method to store data types that are larger than one byte. Words are stored in two consecutive bytes with the low-order byte at the lowest address and the high-order byte at the high address. The same convention applies for 32 bit and 64 bit values.

Following is the detail of the floating-point format:

Short Float (32 bits)

MSB (bit 31) = Sign  
 Bit 30..23 = Exp  
 Bit 22..00 = Mantissa  
 $2^{\text{exp}(-1*\text{bit}22)} + 2^{\text{exp}(-2*\text{bit}21)} \dots$   
 Value = Sign \* 1.mantissa \*  $2^{\text{exp}(\text{EXP}-127)}$

Double Float (64 bits)

MSB (bit 63) = Sign  
 Bit 62..52 = Exp  
 Bit 51..00 = Mantissa  
 $2^{\text{exp}(-1*\text{bit}51)} + 2^{\text{exp}(-2*\text{bit}50)} \dots$   
 Value = Sign \* 1.mantissa \*  $2^{\text{exp}(\text{EXP}-1023)}$

For example, message ID #6, bytes 11..14 (SNR value)(short Float)

byte 11 : 85

byte 12 : AC  
 byte 13 : 41  
 byte 14 : 42  
 short float = 4241AC85  
 Sign = +  
 EXP = 132  
 mantissa = 0.5130773782  
 value = 48.4

## H. MESSAGE STRUCTURE

All the messages have the following form:

byte 1:	SOH
byte 2:	ID # (See Note 1)
byte 3:	Cmpl ID
byte 4:	Message Data Length (0..255)
byte 5 .. n:	n-4 Data bytes
byte n+1 .. n+2:	Checksum

NOTE 1: For transmitter messages, MSB = 0 -> one shot or cancel continuous, MSB = 1 -> continuous unless specified otherwise.



**HOST CPU TO RECEIVER MESSAGES**

## A. MESSAGE SUMMARY

ID	DEFINITION	MESSAGE TYPE	# BYTES
6	Current channel assignment data request	DR	6
20	Navigation data request (user coordinates)	DR	6
22	Ephemeris (ICD-GPS-200 format) request	DR	6
23	Measurement block data request	DR	7
33	Satellite Visibility Data and Status request	DR	6
45	Hardware/Software identification	DR	6
47	Base Station Status request	DR	6
48	Differential Message Status request	DR	6
49	Receiver Status request	DR	6
51	Initiated BIT request	DR	7
63	Initiate link	PM	14
64	Set Channel deselection	CM	16
65	Raw DGPS Data Request	CM	6
80	Set Operating Mode	CM	38
81	Set Mask angle	CM	18
88	Select/Define Datum	CM	38
90	Set SV deselection	CM	18
91	Differential Message Configuration	CM	8
103	Set Date, Time & GPS Time Alignment Mode	CM	21
105	Set default CMC Binary message list	CM	30
110	Configure Main Port Mode	CM	7
112	Switch to Reprogramming Mode	CM	7
<b>LEGEND:</b> CM : Command Message DR : Data Request PM : Protocol Message  Note: Variable length (6 - 94 bytes)			

**B. MESSAGE CONTENT - HOST CPU TO RECEIVER**

MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
6 Current channel assignment data request		This request will cause the receiver to send both messages ID #6 and 7. No data bytes.	N/A	N/A
20 Navigation data request (User coordinates)		No data bytes.	N/A	N/A
22 Ephemeris (ICD-GPS-200 format) request		Each time a new request is sent, the receiver will transmit a complete set of all ephemeris and SV clock data currently acquired. In a case of broadcast mode, the receiver will transmit a complete set and then transmit only on new ephemeris reception. No data bytes.	N/A	N/A
23 10 Hz measurement block data request	5	Request measurement block data for all tracked SV's. The receiver will respond by sending message ID 23 every 100 msec (if requested at 10 Hz). bits 0..1 : Transmission Rate 0 : 1 Hz 1 : 2 Hz 2 : 5 Hz 3 : 10 Hz 2..7: Reserved (shall be 0)	N/A	N/A
45 Software Identification request		No data bytes.	N/A	N/A
47 Base Station Status request		No data bytes.	N/A	N/A
48 Differential Message Status request		No data bytes.	N/A	N/A
49 Receiver Status request		No data bytes.	N/A	N/A
51 Initiated BIT request	5	0 = PowerUp BIT Results 1 = Initiate a Customer BIT 2-255 = Reserved	N/A	N/A
63 Initiate Link	5..12	This request will cancel all previous data request messages within 2 seconds. Password (UGPS-000), in ASCII format, U character first	N/A	char [8]
64 Set Channel deselection		Set deselection criteria for all 12 channels if password valid. The channels to be deselected should be indicated in a bit map form. 1 in the bit map specifies that the corresponding channel shall be deselected.	N/A	N/A

MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
64 (Cont'd)	5..12 13 14	Password (UGPS-000), in ASCII format, N/A U character first bit map (bit 0 -> ch #1, bit 7 -> ch #8) bit map (bit 0 -> ch #9, bit 3 -> ch #12)		char [8]
80 Set Operating Mode	1-4  5-12          13-20  21-28  29-36  37-38	CMC Header  Password (UGPS-XXX), in ASCII format, U character first. where XXX: 000        -    Set User Position (AllStar compatible) R00        -    Force to Rover Mode (position        not saved) GSP        -    Get Survey Position BYY        -    Set Base Position and Base Information SYY        -    Force to Survey Mode where YY: bytes 11..12 (Station ID and Station Health)  bits 0..9 :    Station ID (10 bits:1-1023) bits 10..12 : Station Health(as per RTCM) bits 13..15 : Reserved Interpreted field [000 BYY ]    Altitude Ellipsoid [SYY]        Survey time [0.0..48.0] [R00 GSP]    Don't Care Interpreted field [000 BYY ]    Latitude [SYY R00 GSP] Don't Care Interpreted field [000 BYY ]    Longitude [SYY R00 GSP] Don't Care Checksum	N/A  N/A	N/A  char[8]
81 Set Mask angle	5..8 9..16	Mask angle (0 .. $\pi/2$ , short float) Reserved The value will be stored in NVM.	N/A	N/A

MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
88 Select/Define datum	5	Select the datum used to report the position and define user-defined datum. Function 0 : Select datum 1 : Define a user-defined datum 2 : Select and define a user-defined datum	N/A	N/A
	6	datum number (from 0 to 63, see Supported Datum List, Appendix TBD)	N/A	N/A
	7,8	dx	Meter	signed short
	9,10	dy	meter	signed short
	11,12	dz	meter	signed short
	13..20 29..36	a (semi-major) Reserved Note: The navigation data (user coordinates) message contains the datum currently in use.	meter	long float
90 Set SV deselect	5..12	Set deselection for all 32 SVs if password valid. The SVs deselect is indicated in a bit map form. 1 in the bit map specifies that the corresponding SV shall be deselected. Password (UGPS-000), in ASCII format, U character first	N/A	char [8]
	13	bit map (bit 0 → SV #1, bit 7 → SV #8)	N/A	N/A
	14	bit map (bit 0 → SV #9, bit 7 → SV #16)	N/A	N/A
	15	bit map (bit 0 → SV #17, bit 7 → SV #24)	N/A	N/A
	16	bit map (bit 0 → SV #25, bit 7 → SV #32)	N/A	N/A
91 Differential Message Configuration	5	Message type and protocol bits 0..5: 0: Clear All Messages 1-63: Message Type bits 6,7: 00: RTCM 01: Reserved 10: RTCA 11: Reserved	N/A	N/A
	6	Rate 0: Stop transmitting 1-255: every xx second(s)	seconds	byte
103 Set Date Time & GPS Time Alignment Mode		Enter the date and time (UTC). This data is accepted only if a SV is not presently being tracked and if password is valid.	N/A	N/A

MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
103 (Cont'd)	2	bit 7 0-Valid Time, 1-Invalidate the internal time	N/A	N/A
	3	bit 7 1-Valid Time, 0-Invalidate the internal time	N/A	N/A
	5..12	Password, in ASCII format, U character first UGPS-000: the date and time parameter will be applied UGPS-001: the date and time parameter won't be applied but will force the receiver to align its measurements (and TIMEMARK signal) on GPS time after the next power-up. A master reset is requested 10 seconds after the acknowledge of the message ID #103 to ensure the proper operation of the time alignment function. Contact CMC for more information. UGPS-002: The date and time parameter won't be applied but will force the receiver to not align its measurements (and TIMEMARK signal) on GPS time.	N/A	char [8]
	13..15	UTC Time resolution: 1 second	HR:MN:S C	byte:byte: byte
	16..19	Data resolution: 1 day	DY:MO:Y R	byte:byte: byte
105 Set Default CMC Binary Message List	5	bit 0: Reserved bit 1: Message ID#1 Flag: 0 : won't be transmitted 1 : will be transmitted bit 2: Message ID#2 Flag: 0 : won't be transmitted 1 : will be transmitted bit 3-7: Message ID#3-7 Flags: 0 : won't be transmitted 1 : will be transmitted	N/A	N/A
	6..20 21..28	Message ID #8-127 Flags Reserved	N/A	N/A
110 Configure Main Port Mode	5	bits 0-6: Baud Rate (in 300 bauds unit, 1=300, ... 32=9600, ... 64=19200, 65=38400) bit 7: Mode : 1 = CMC Binary, 0 = NMEA	N/A	N/A

MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
112 Switch to Reprogramming Mode	5	Baud Rate (1=300, ... 32=9600, ... 64=19200)	300 bauds	N/A

## RECEIVER TO HOST CPU MESSAGES

### A. MESSAGE SUMMARY

ID	DEFINITION	MESSAGE TYPE	RATE (SEC)	# BYTES
6	Current channel assignment data (1-6)	UR/FR	1	91
7	Current channel assignment data (7-12)	UR/FR	1	91
20	Navigation data (user coordinates)	UR/FR	1*	77
22	Ephemeris (ICD-GPS-200 format) data	UR/FR	1	79
23	Measurement block data	UR/FR	VAR	149
33	Satellite visibility data and status	UR/FR	1*	67
45	Hardware/Software identification data	UR	1	101
47	Base Station Status data	DR	1	50
48	Differential Message Status data	DR	1	29
49	Receiver Status data	DR	1	12
51	Initiated BIT result	UR	1	40
63	Initiate link	PM	0.1	14
65	Raw DGPS Data	PM	VAR	VAR
125	Link overload error message	PM	1	22
126	Acknowledge message	PM	0.1	11
<p><b>LEGEND:</b>    CM : Command Message                                       DR : Data Request                                       PM : Protocol Message</p> <p>Note for PM and SM: The protocol messages (PM) and status messages (SM) are scheduled to be output once per second or per 100 msec.</p> <p>* Note:        Transmitted twice per second when in 2 Hz PVT mode</p>				

### B. MESSAGE CONTENT - RECEIVER TO HOST CPU

MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
6 Current channel assignment data (1-6)	5	Data set number Channel 1 assignment data	N/A	N/A
	6	SV # and type bit 0 .. 4: SV # (0..31) bit 5 .. 7: Reserved	N/A	N/A
	7..10	Carrier frequency resolution..cycles	cycles	unsigned 32
	11..14	SNR	dB-Hz	short float
	15..18	Reserved		
	19	Status Bit 0-1 encodes tracking state 00 -> not ready 01 -> bits ready 10 -> meas ready 11 -> failed Bit 2-3 encodes allocation state 00 -> idle, 01 -> location, 10 -> tracking Bit 4 encodes channel mode, 1 -> automatic, 0 -> manual	N/A	N/A
	20..33	Channel #2 assignment data	as per ch.1	as per ch.1
	34..47	Channel #3 assignment data	as per ch.1	as per ch.1
	48..61	Channel #4 assignment data	as per ch.1	as per ch.1
	62..75	Channel #5 assignment data	as per ch.1	as per ch.1
76..89	Channel #6 assignment data	as per ch.1	as per ch.1	
7 Current Channel Assignment Data (7-12)	5	Data set number	N/A	N/A
	6..89	Channel 7-12 assignment data	as per ch.1	as per ch.1
20 Navigation Solution		The message is output once per second upon reception of a message ID #20 request.	N/A	N/A
	5-14	The latency on this message is less than 0.5 seconds. The latency defined here refers to the time difference between the time tag of the computed position and the time of transmission of the first message byte. UTC Time 5 [bit 0..4]-> hour in day 6->minute in day 7-14->seconds in day 5[bit5] -> 0: time not corrected by UTC parameters 1: time corrected by UTC parameters	hour minute seconds N/A	byte byte double N/A

MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
20 (Cont'd)	15-18	Date 15[bit 0..6] -> Day of the year 16 -> Month of the year 17-18 -> year	day month year	byte byte word
	19-26	Latitude	radians	double
	27-34	Longitude	radians	double
	35-38	Altitude	meters	float
	39-42	Clock Bias	meters/ sec	float
	43-46	Clock Drift	radians	float
	47-50	Velocity North	meters/ sec	float
	51-54	Velocity East	meters/ sec	float
	55-58	Velocity Up	meters/ sec	float
	59-62	HFOM	meters	float
	63-66	VFOM	meters	float
	67-70	GPS Time [0..604800]	meters	float
	71	bits 0-4: NAV Mode 0 -> Init. Required 1 -> Initialized 8 -> BaseStation 9 -> CodeNav 10 -> RTK	N/A	N/A
	72	bits 0..3 : Number of SVs used in the solution bits 4..7 : Reserved	N/A	N/A
	73	Reserved	N/A	N/A
	74-75	Week Number	N/A	N/A
	76-77	Checksum	N/A	N/A
22 Ephemeris Data	5	This message contains information for one Satellite ephemeris data. It is transmitted at a rate of one message per second until all ephemeris data list completed and then transmit only if new ephemeris occurs. The user is directed to ICD-GPS-200 for specifics on the format of the ephemeris data. bits 0..4 : SV Number bits 5..7 : reserved	N/A	N/A
	6..77	Ephemeris sub-frame 1-3/words 3-10 MSB of byte 6 is the Bit 61 of subframe 1	N/A	N/A



MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
23 10 Hz Measurement Block Data	5-6	Reserved	N/A	N/A
	7	Number of measurement blocks (N)	N/A	N/A
	8..15	Predicted GPS Time	double	seconds
	16	bits 0..5 : SV # (0..31) bit 6 : reserved bit 7 : Toggle at each Ephemeris Transmission	N/A	N/A
	17	SNR	0.25 dBHz	unsigned char
	18..21	Code Phase (1/1024 half chip) range : 0 .. 2095103999	1/1024 half chip	unsigned long
	22..25	Integrated Carrier Phase bit 0-1 : 0 : Ready 1 : Phase Unlock 2 : Cycle Slip Detected 3 : Not Ready	N/A	N/A
		bits 2-11 : Carrier Phase (0-1023)	1/1024 cycle cycles	
	26	bits 12-31: Integrated Number of Cycles range: natural roll over Cycle_Slip Counter Increment by 1 every time a cycle slip is detected during a 10ms period range: natural roll over Measurement block #2 . . Measurement block #N	cycle slip	unsigned char
			as per block #1	as per block #1
33 Satellite Visibility Data and Status	5	bit 0...3: Total number of Satellites in view bit 4..7: reserved Data transmission of up to 12 satellites in view listed in decreasing elevation order. Satellite visibility data of the 1 <sup>st</sup> SV:	N/A	N/A
	6	Computed data bit map bit 0..4 : SV Number bit 5..6 : SV Status 0 = In View 1 = Tracking 2 = MeasReady 3 = Used by Nav	N/A	N/A
	7	bit 7 : Differential Corrections available Elevation range : -90..90	degree	signed char
	8-9	Azimuth range : 0..360	degree	word
	10	bits 9-15 : Reserved SNR range : 0..90	dB	byte

MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
33 (Cont'd)	11..15	Satellite visibility data of the 2 <sup>nd</sup> SV	as per SV1	as per SV1
	16..20	Satellite visibility data of the 3 <sup>rd</sup> SV	as per SV1	as per SV1
	21..25	Satellite visibility data of the 4 <sup>th</sup> SV	as per SV1	as per SV1
	26..30	Satellite visibility data of the 5 <sup>th</sup> SV	as per SV1	as per SV1
	31..35	Satellite visibility data of the 6 <sup>th</sup> SV	as per SV1	as per SV1
	36..40	Satellite visibility data of the 7 <sup>th</sup> SV	as per SV1	as per SV1
	41..45	Satellite visibility data of the 8 <sup>th</sup> SV	as per SV1	as per SV1
	46..50	Satellite visibility data of the 9 <sup>th</sup> SV	as per SV1	as per SV1
	51..55	Satellite visibility data of the 10 <sup>th</sup> SV	as per SV1	as per SV1
	56..60	Satellite visibility data of the 11 <sup>th</sup> SV	as per SV1	as per SV1
	61..65	Satellite visibility data of the 12 <sup>th</sup> SV	as per SV1	as per SV1
45 Software Identification Information	5..18	Operational S/W Part number (XXX-XXXXXX-XXX)	N/A	char [14]
	19..36	Reserved ASCII string	N/A	char [18]
	37..50	Boot S/W Part number (xxx-xxxxxx-xxx)	N/A	char [14]
	51..90	Reserved	N/A	N/A
	91..94	Boot Checksum	N/A	N/A
	95..98	Operational Checksum	N/A	N/A
	99	Reserved	N/A	N/A
47 Base Station Status	5	This message is output once per second upon reception of a message ID #47 request.	N/A	N/A
		BaseStatus bits 0-1 : Base Status 0 : Not in Base 1 : Position Not Initialized 2 : Base Initialized 3 : Reserved bits 2-4 : Baud Rate 0-300   1-600   2-1200   3-2400 4-4800   5-9600   6-19200   7-38400 bits 5-7 : Reserved	N/A	N/A
	6-13	Time Remaining Survey	hours	double
	14-17	Base Station Position CEP	meters	float
	18-25	Base Station Position Latitude	radians	double
	26-33	Base Station Position Longitude	radians	double
	34-41	Base Station Position Height	meters	double
	42-45	Reserved	N/A	N/A
	46	Number of Differential Message bit 0-4 : Number of Differential Message bit 5-8 : Reserved	N/A	byte
	47-48	byte 1 : Msg Type	N/A	byte
	49-50	byte 2 : Programmed Msg Rate Period	sec	byte
		Checksum	N/A	N/A
48 Differential Message Status	5	This message is output at a nominal rate of once per second upon reception of a message ID #48 request.	N/A	byte
		Station Id # (bits 0..7)		

MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
48 (Cont'd)	6	bit 0-1 : Reserved bit 2-4 : Station Health bit 5-6 : Station Id bit 8-9 bit 7 : Reserved	N/A	byte
	7-14	Msg Type # Received (Bitmap: bit0 = 1, bit63 = 64)	N/A	N/A
	15..16	Receiver Mode Differential data link - Valid Word Count Base Mode Reserved	N/A	N/A
	17..18	Receiver Mode Differential data link - Parity Error Count Base Mode Reserved	N/A	N/A
	19	Reserved	N/A	N/A
	20	Reserved	N/A	N/A
	21	Reserved	N/A	N/A
	22	Reserved	N/A	N/A
	23	Reserved	N/A	N/A
	24	Reserved	N/A	N/A
	25..26	bits 0..12: ZCount of last message 1, 2, 3, 9, or 59 Receiver Mode bits 13..15: DGPS Status 0 -> DGPS Disabled 1 -> Initialization/Synchronization 2 -> Correcting 3 -> Bad GDOP 4 -> Old corrections 5 -> Station unhealthy 6 -> Too few SVs 7 -> Reserved Base Mode Reserved	N/A	N/A
	27..28	Reserved	N/A	N/A
	29..30	Checksum	N/A	N/A
49 CPU - Receiver Status		This message is output at a nominal rate of once per second upon reception of a message ID #49 request.	N/A	N/A
	0	System Status 1 bit Failure=1 Normal=0	N/A	N/A
	1-2	Last Power Mode 2 bits 0 : no NVM 1 : Cold Start 2: Warm Start	N/A	N/A
	6-7	Almanac Week Number	week	N/A
	8	Mask Angle	0.01°	unsigned word
	9	bit 0 : Tropo Model Enabled bit 1 : Mean Sea Level Enabled	N/A	N/A
	10-11	TCXO Drift	Hz	signal word
	12-13	TCXO Ageing	0.1 ppm	unsigned char
	14	NAV mode	N/A	N/A

MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
49 (Cont'd)	15	bit 0 : COM1 Overflow bit 1 : COM2 Overflow	N/A	N/A
	16	CPU Usage	N/A	N/A
51 Initiated BIT Result	5	bit 0-7 : Copy of the Initiated BIT request message byte 1	N/A	N/A
	6	General Results (0=fail, 1=Pass) bit 0 : RAM bit 1 : Flash bit 2 : Eeprom bit 3 : Uart bit 4 : Real Time Clock bit 5 : Correlator & RF bit 6-7 : Reserved	N/A	N/A
	7-9	Reserved	N/A	N/A
	10	Memory Test Results (0=ok, 1=failure) bit 0 : Bad Boot S/W Checksum bit 1 : Bad Operational S/W Checksum bit 2-4: FLASH Error Code if different of 000 : Receiver can not be reprogrammed bit 5-7 : Reserved	N/A	N/A
	11	EEPROM Status bit 0-7 : Number of Usable Pages	N/A	N/A
	12	Primary Port (UART) busy bit 0 : UART not ready or UART busy bit 0 : UART not ready or UART busy bit 1 : TX not full flag error bit 2 : No Data received during internal loop tests bit 3 : Framing or Parity error bit 4 : RX not full flag error bit 5 : OVERRUN test failed bit 6-7 : Reserved	N/A	N/A
	13	Auxiliary Port (UART) results bit 0 : UART not ready or UART busy bit 1 : TX not full flag error bit 2 : No Data received during internal loop tests bit 3 : Framing or Parity error bit 4 : RX not full flag error bit 5 : OVERRUN test failed bit 6-7 : Reserved	N/A	N/A
	14	RTC results bit 0-2 : RTC warning bit 3-4 : Data Retention register error bit 5-7 : Reserved	N/A	N/A
	15	RF Test Results 0-7 : RF warning code	N/A	N/A

MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
51 (Cont'd)	16	Global Correlator test results #1  bit 0 : Channel 0 error in I&Q test bit 1 : Channel 1 error in I&Q test bit 2 : Channel 2 error in I&Q test bit 3 : Channel 3 error in I&Q test bit 4 : Channel 4 error in I&Q test bit 5 : Channel 5 error in I&Q test bit 6 : Channel 6 error in I&Q test bit 7 : Channel 7 error in I&Q test	N/A	N/A
	17	Global Correlator test results #2 bit 0 : Channel 0 error in I&Q test bit 1 : Channel 1 error in I&Q test bit 2 : Channel 2 error in I&Q test bit 3 : Channel 3 error in I&Q test bit 4 : Channel 4 error in I&Q test bit 5 : Channel 5 error in I&Q test bit 6 : Channel 6 error in I&Q test bit 7 : Channel 7 error in I&Q test	N/A	N/A
	18	Global Correlator test results #3 bit 0 : Channel 9 error in I&Q test bit 1 : Channel 10 error in I&Q test bit 2 : Channel 11 error in I&Q test bit 3 : Channel 12 error in I&Q test bit 4 : Channel 9 error in Measurement test bit 5 : Channel 10 error in Measurement test bit 6 : Channel 11 error in Measurement test bit 7 : Channel 12 error in Measurement test	N/A	N/A
	19-30	Reserved		
	31-40	Reserved		
63 Initiate Link	5..12	Password (UGPS-xxx), in ASCII format, U character first	N/A	char [8]
65 Raw DGPS Data	5	Control byte Bits 0..3 : Sequence Number (0..15) bits 4..7 : Protocol 0 : RTCM 1 : RTCA 2..15 : Reserved	N/A	N/A
	6..n	Raw DPGS data	N/A	N/A
125 Link Overload Error Message	5..20	bit map (bit 0 -> ID #1, bit 127 -> ID #127)	N/A	N/A

MESSAGE	BYTE	DESCRIPTION	UNIT	TYPE
126 Acknowledge Message	5	ID of first message acknowledged	N/A	N/A
	6	ID of second message acknowledged	N/A	N/A
	7	ID of third message acknowledged	N/A	N/A
	8	ID of fourth message acknowledged	N/A	N/A
	9	ID of fifth message acknowledged	N/A	N/A

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## **SECTION VI - PRODUCT TEST AND QUALITY ASSURANCE PROVISIONS**

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## SECTION VI

### PRODUCT TEST AND QUALITY ASSURANCE PROVISIONS

#### QA CONFORMANCE TESTING

The equipment is subjected to testing in accordance with this section to demonstrate compliance with this specification.

Production tests are those tests which are conducted on each production equipment prior to delivery.

#### STANDARD TEST CONDITIONS

Unless otherwise specified, the equipment is subjected to the acceptance tests under the following conditions:

- |                |   |   |
|----------------|---|---|
| 1. Temperature | - | Room Ambient +25 deg. C $\pm$ 10 deg. C |
| 2. Altitude    | - | Normal Ground                           |
| 3. Vibration   | - | None                                    |
| 4. Humidity    | - | Room Ambient                            |

#### USER-DEFINED TESTS

The user is encouraged to design a customized test to ensure his system functions properly.

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## SECTION VII - SERVICE AND SUPPORT

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## SECTION VII

### SERVICE AND SUPPORT

#### POINTS OF CONTACT - CMC

**Postal Address:**

CMC Electronics Inc.  
Components Division  
GPS OEM Group, Box 92  
600 Dr.-Frederik-Philips Boulevard  
St-Laurent, QC, CANADA  
H4M 2S9

**WEB Site:**

[http:// www.cmcelectronics.ca](http://www.cmcelectronics.ca)

**Marketing / Sales:**

Tel : 514 - 748 - 3070  
Fax : 514 - 748 - 3017  
Email : [gpsmarket@baesystems-canada.com](mailto:gpsmarket@baesystems-canada.com)

**Contracts / PO / Shipment Status:**

Tel : 514 - 748 - 3000 Ext 4943  
Fax : 514 - 748 - 3017  
Email : [gpscontract@baesystems-canada.com](mailto:gpscontract@baesystems-canada.com)

**Technical Support:**

Tel : 514 - 748 - 3070  
Fax : 514 - 748 - 3130  
Email : [gpshelp@baesystems-canada.com](mailto:gpshelp@baesystems-canada.com)

**FTP Site:**

<ftp.baesystems-canada.com>

#### SERVICE AND REPAIRS

All receivers conform to the specifications stated herein. Should any damage occur to the receivers during shipping, handling, or misuse by the user, CMC can service them. Try to be as complete and accurate as possible when you describe a problem.

**PRODUCT UPDATED**

All product updates will be advertised on our Web site.

**TROUBLESHOOTING AND FREQUENTLY ASKED QUESTIONS (FAQ)**

A FAQ list is available on our Web site.

**CONSULTATION**

Technical consultation can be obtained from CMC if GPS expertise is needed for the integration of the receiver into your application. We can provide support either at the system design, implementation, or testing phase. For more details, please contact Technical Support (refer to Points of Contact section above).

## APPENDIX A - RECEIVER DEVELOPMENT KIT

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## APPENDIX A

### RECEIVER DEVELOPMENT KIT

#### OVERVIEW

The Receiver Development Kit allows new users to easily evaluate the GPS receiver. The Development Kit implements the receiver control operation and I/O functions of the receiver using an IBM-compatible personal computer (PC), a serial port, an external geodetic GPS antenna, and an I/O cable with a 115 VAC to 12 VDC power adapter. The GPS receiver is contained in an aluminum extrusion unit called STARBOX, with I/O connectors and status LEDs.

STARVIEW is an MS-Windows application running on a PC that allows communication with the receiver. All commands and data requests can be sent through this application and all received data is decoded and displayed in specific windows. A data logging facility is also provided within this tool. Details on the use of STARVIEW is provided in the STARVIEW User's Manual (Ref [5]).

This Appendix explains how to configure the Development Kit and the receiver, and how to interconnect the equipment.

#### DESCRIPTION

The Development Kit contains the following equipment:

- a GPS receiver contained in the STARBOX unit
- 115 VAC to 12 VDC adaptor
- RS-232 I/O cables (DB-25 to DB-9 connectors and lighter plug)
- STARVIEW software
- RT•STAR User's Manual
- STARVIEW User's Manual

Your selection of Geodetic GPS antenna among the following choices

- Choke Ring Antenna with +12db, +26dB or +36 dB gain
- Geodetic 18 cm dia. ground plane with +12db, +26dB or +36 dB gain
- Airborne Geodetic with +12db, +26dB or +36 dB gain

Your selection of low loss RF Coax Cables with length from 1meter up to 99 meters.

Furthermore, the following equipment is also required:

- an IBM-compatible PC with MS-Windows'95 or higher, or MS-Windows NT 3.51 or higher; a portable PC is highly recommended for tests outdoors
- color monitor with SVGA

Note: For exact CMC part numbers and availability, please refer to the latest price list (refer to Section VII).

## EQUIPMENT INTERCONNECTION

The provided equipment should be interconnected as per Figure A-1.

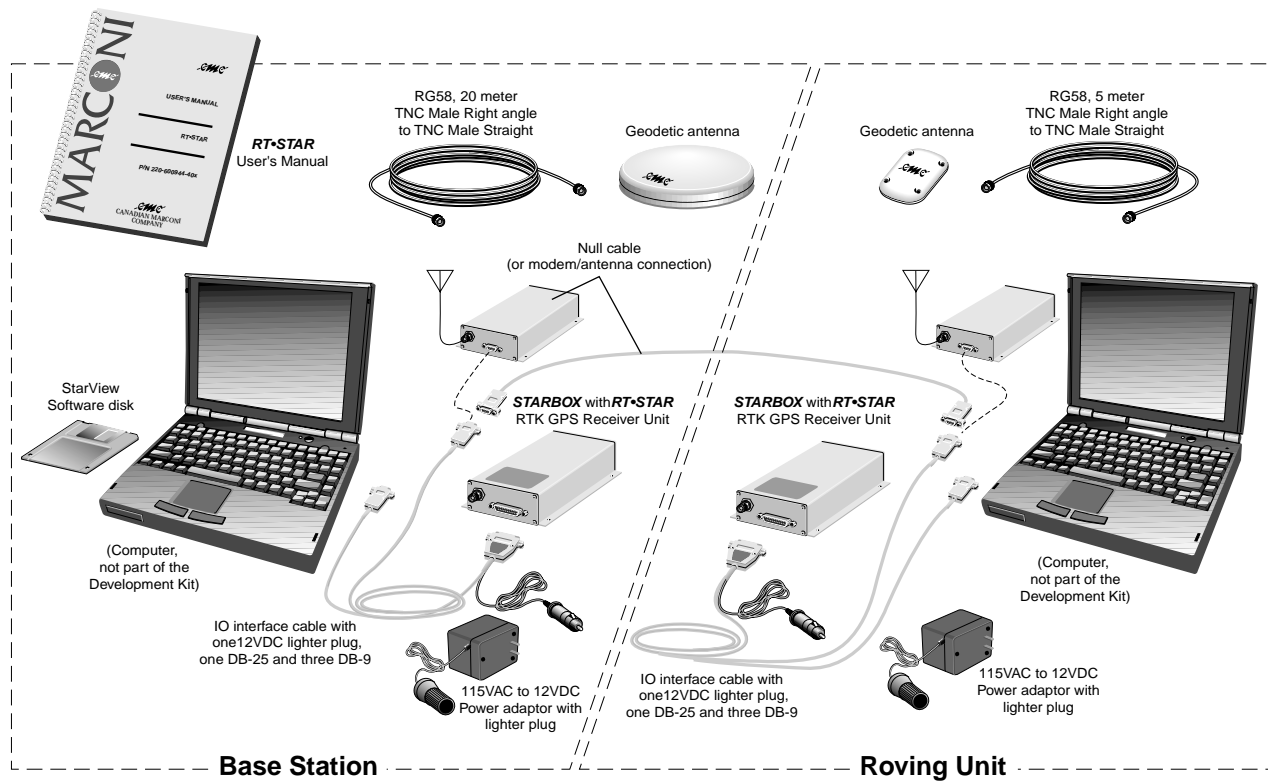


Figure A-1. Receiver Equipment

## APPENDIX B - STARBOX

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## APPENDIX B

### STARBOX

#### OVERVIEW

The STARBOX is a robust metal casing that houses the GPS receiver and a power supply. Communication with the receiver is performed via a DB-25 connector. Figure B-1 depicts the interconnection required for the STARBOX.

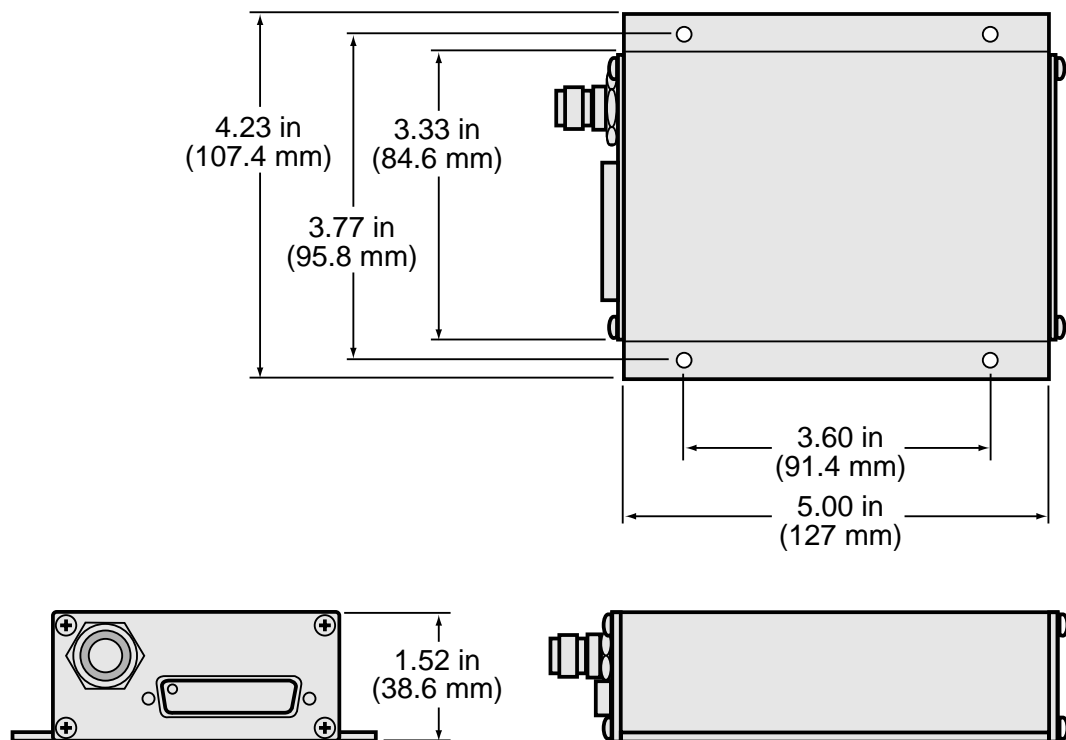


Figure B-1. Outline Drawing

#### DESCRIPTION

The pinout of the DB-25 connector is as follows:

DB-25 PIN #	STAR-BOX	DESCRIPTION
1	DISC_IP_1	OPEN-GND CMOS discrete input with 10K pull-up resistor (Note 3)
2	GND	Ground Connection
3	DISC_IP_2	OPEN-GND CMOS discrete input with 10K pull-up resistor (Note 3)
4	RESERVED	
5	TIMEMARK_1	1 Pulse Per Second Output TTL level with a 100 series resistor
6	Serial Intf TX 1	Main Serial Interface Port Transmitter (Note 1)
7	Serial Intf RX 1	Main Serial Interface Port Receiver (Note 2)
8	GND	
9	RESERVED	
10	RESERVED	
11	TEST	CMOS discrete input with 10K $\Omega$ pull-up resistor Force the Reprogramming Mode at power up if connected to Ground. Shall be left OPEN for normal operation
12	RESERVED	
13	GND	
14	RESERVED	
15	RESERVED	
16	DISC_OP_1	CMOS discrete output with 100 $\Omega$ series resistor
17	RESERVED	
18	TIMEMARK_2	OPEN-DRAIN output with a 10K $\Omega$ pull-up resistor 500ma maximum
19	GND	
20	Serial Intf TX 2	Auxiliary Serial Interface Port Transmitter (Note 1)
21	Serial Intf RX 2	Auxiliary Serial Interface Port Receiver (Note 2)
22	RESERVED	
23	RESERVED	
24	RESERVED	
25	12V_DC	9V-36V Power Input With Reversed Voltage Protection

Note 1 : RS232 Driver  
 $\pm 5V$  in 3K $\Omega$   
 Short Circuit Protection 60mA max

Note 2 : RS232 Receiver  
 $V_{il}$  0.7V  
 $V_{ih}$  2.0V  
 $\pm 8V$  max for normal operation

Note 3 : Should be left OPEN in not used in the application

## RF CONNECTOR

The GPS RF connector is a TNC female connector.

## LEDs

The STARBOX has 2 LEDs:

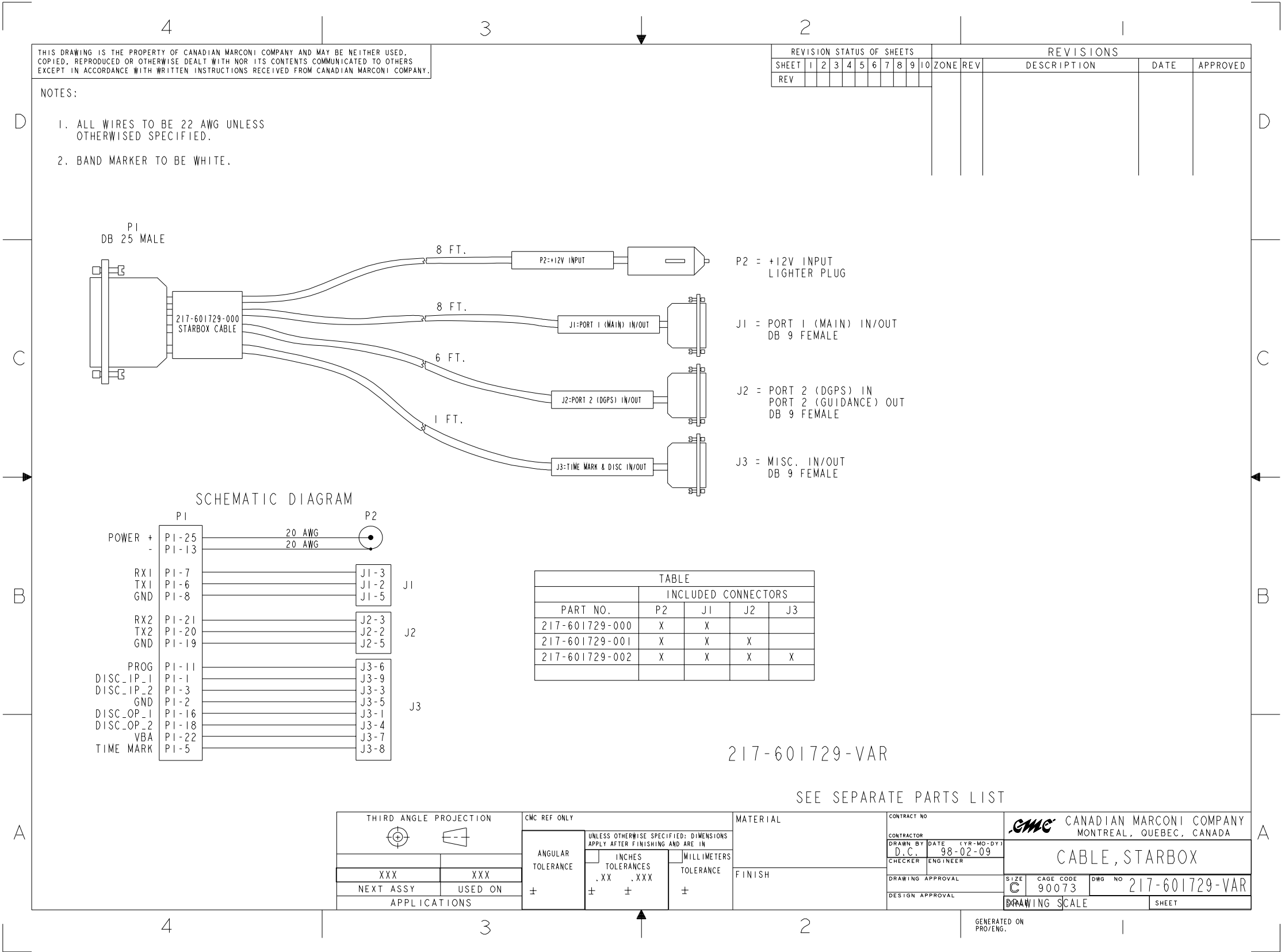
LED's COLOR	DESCRIPTION
YELLOW	When flashing at a 1 Hz rate, indicates that the receiver had enough satellite information to perform a Navigation solution. Doesn't flash by default after a power-up.
GREEN	Valid Internal 5 Volt Indicator

## CABLE

A schematic of the cable to be used with the STARBOX is depicted in Figure B-2. It's part number is-217-601729-VAR.



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## APPENDIX C - EXTERNAL INTERFACE CHARACTERISTICS

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I/O ELECTRICAL CHARACTERISTICS.....	C-2

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## APPENDIX C

## EXTERNAL INTERFACE CHARACTERISTICS

## CONNECTOR PIN ASSIGNMENT

Figure C-1 shows the Interface and Power connector (J1 or J3 depending of OEM variation) pin assignment.

J1 PIN #	J3 PIN #	SIGNAL NAME	I/O	COMMENTS
1		Reserved		
2		Disc_IP_3	I	Note 4
3		Reserved		
4	1	Reserved		
5	3	Power Control Input	I	Note 4
6	4	Rx_No_3 (Optional Port)	I	
7	5	Time Mark (1 PPS) Output	O	
8		Ground		
9	6	Disc_IP_1	I	Note 2,3,5
10	2	Tx No 3 (Optional port)	O	
11	7	Rx No 2 (Auxiliary port)	I	Note 4
12	9	Ground		
13	8	Tx No 2 (Auxiliary port)	O	
14	10	Disc_IP_2	I	Note 4
15	11	VDD		
16		Ground		
17	12	Rx No. 1 (Primary port)	I	Note 5
18	13	Ground		
19	14	Tx No. 1 (Primary port)	O	
20	15	Reserved		
21	17	+5V Digital		
22		Ground		
23	16	Preamplifier (Active Antenna supply)		
24	18	Ground		
25	20	Disc_I/O_1	I/O	Note 5
26	19	+5V RF		

Figure C-1. J1 and J3 Interfaces and Power Connector Pin Assignment

- Note 1: Not used.  
 Note 2: For normal operation the pin should be tied to GND (preferred option) or left open.  
 Note 3: The pin has been reserved for the reprogramming mode (see Appendix G).  
 Note 4: On-board pull-up resistor  
 Note 5: On-board pull-down resistor  
 Note 6: Reserved pins shall be left unconnected

**I/O ELECTRICAL CHARACTERISTICS**

Figure C-2 shows the voltage level limits for all different I/O signals:

SIGNAL NAME	TYPE	Vil max	Vih min	Vol max	Voh min	Input Rise & Fall Time
		Volt	Volt	Volt	Volt	uSEC
Power Control Input (note1)	I	0.50	2.00			< 1
Disc_IP_1,Disc_IP_2, Disc_IP_3,DISC_IO_1, Rx No.1 ,Rx No 2,	I	0.7	2.3			< 1
Rx No 3	I	0.8	2			
Tx No 1, TX No 2, Timemark Output 1PPS, DISC_IO_1	O			0.4	$(0.8 * VDD) - 0.1$ $I_o \leq 200\mu A$	
Tx No 3	O			0.4	2.4	

Note 1: A LO pulse of 150ns minimum will invoke a master reset to the receiver.

Note 2: Conditions : 5V +/- 5%(for all limits)

Figure C-2. I/O Signals Voltage Limits

**APPENDIX D - ANTENNA SPECIFICATIONS****CONTENTS**

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## APPENDIX D

### ANTENNA SPECIFICATIONS

The GPS antenna is an important part of the total system performance and should be selected depending of your application. All the GPS receivers designed and manufactured by CMC in the GPS-OEM, include an Low Noise Amplifier (LNA) before the the RF ASIC. This +20dB LNA permit raisonnaable performances with a passive GPS antenna. But depending of the cable loss between the antenna element and the GPS receiver and also the position accuracy requirements, then a +12dB up to +36dB Active GPS Antenna could be needed.

This appendix is divided in characteristics for high end Active Geodetic Antenna including Choke Ring Antenna, then lower cost Active Antenna and then Passive Antenna. CMC does not manufacture GPS Antenna but because of the high volume consume by our subsidiary, the GPS -OEM group is able to offer on the re-sell market very good antenna at a very competitive price. There are many GPS suppliers around the world, CMC tried most of them and selected AeroAntenna Technologies Inc. as one reputable source of its antenne supply. This is not to say that any other GPS Antenna supplier will not perform well with our receivers. It is the user responsibility to select the GPS Antenna which best full fill its requirements.

CMC is also able to offer the coax cables required between the GPS Antenna and the our Receiver. You will also find in this section, different coax cables required in your GPS system. The end of this section includes very detailed Antenna drawings

#### CABLE SELECTION

The interconnection cable between the GPS Antenna and the Receiver is of prime importance for the proper performance of the system. Three parameters are to be considered: the Loss, Isolation, and Outer Diameter. The bigger the Outer Diameter, the lower the Loss. The Loss increases with the length of the cable and decreases with extra isolation. If the highest accuracy possible is not required CMC GPS-OEM receivers can accept a total cable loss of 3 dB. Depending of the cable type, this could represent a cable run from 2 meters up to 10 meters if expensive cable is used.

Table D-1 details the specifications of the RG-58 Low Loss Cable (RG-58/U LLDS80) used in the CMC GPS Antenna Cable 217-601730-XXX. The RG-58/U LLDS80 is a custom-made low loss noise coax cable made according to AeroAntenna specifications. It is a double-shielded cable similar to Belden Type 9310 but with the improvement of having 85% miniumum coverage of the second shield versus the Belden at 55% coverage. The electrical characteristics are included in Table D-1.

Center Conductor	#20 Bare Copper wire, Resistance - 33.1 ohms per Km
Insulation	Polyethylene
Inner Shield	Aluminium Foil - 100% coverage
Outer Shield	Tinned copper braid - 85% coverage, Resistance 45.9 ohms per Km
Jacket	Black PVC
Nominal Impedance	50 ohms
Nominal Vel. of propagation	66%
Nominal Capacity	101.7 pf per meter
Attenuation	@ 1000mhz: 44.3 dB per 100 meters (or 54 dB @ 1575MHz)

Table D-1. Coax Cable Specifications

Table D-2 shows the minimum and maximum cable length when using CMC GPS Antenna cable drawing 217-601730-XXX used in conjunction with the smaller cable (CMC Drawing 217-601727-XXX) which is usually required between the receiver and the chassis case of the user system. You will find the drawings for these two cables at the end of this section.

CMC Cable PNs	ANTENNA GAIN	CABLE TYPE	MAX. LENGTH *	MIN. LENGTH
217-601730-XXX	0 dB (no LNA)	RG-58 Low Loss	3 meter (3 dB)	0 feet
217-601730-XXX	+12dB	RG-58 Low Loss	20 meter (12dB)	0 feet
217-601730-XXX	+26dB	RG-58 Low Loss	50 meter ( 28dB)	20 meter (12dB)
217-601730-XXX	+36dB	RG-58 Low Loss	65 meter (36dB)	50 meter (28dB)

Table D-2. Antenna Gain Depending on Cable Length Required

- \* A 1 dB loss for the coax cable is usually required between the RG-58 cable and the GPS Receiver MCX connector and it is included in attenuation number in parentheses. If the distance between the antenna and the GPS receiver needs to be longer than 65 meters, the user shall select an other type of coax cable with a lower lost per meter.

The +26dB and +36dB antenna can accept a supply voltage between 5V and 18VDC. It is recommended to compute the drop in the coax cable so the active antenna will always see the minimum operating voltage of 4.5Volt. The Table D-3 list the current taken by each of these antenna.

Antenna Gain	Current Consumption
+12dB	20 mA
+26dB	35 mA
+36dB	50 mA

Table D-3. Typical Current Consumption Versus Antenna Gain

**GEODETTIC ACTIVE ANTENNA**

For RTK applications where centimeter-level accuracy is required, it is strongly recommended to use an active geodetic GPS antenna if possible. In the event where the cable length between the receiver and the antenna is very short (less than one meter), a passive antenna could then be considered. Table D-4 lists the specifications for recommended Passive Antennas. Complete drawings could be find at the end of this appendix.

Antenna Types	CMC Part Numbers	SUPPLIER Part Numbers
Choke ring antenna with trypod mount and permanent mount. Ground plane included	201-990146-888	AT575-90W with +12 dB
	201-990147-607	AT575-90W with +26 dB
	201-990147-680	AT575-90W with +36 dB
Completely sealed round disk antenna with 1 inch tread and 5/8 inch adaptor, with build in ground plane	201-990146-887	AT575-75W with +12dB
	201-990147-606	AT575-75W with +26 dB
	201-990147-679	AT575-75W with +36dB
Smaller mobile mount, ground plane required	201-990147-684	AT575-32W with +12dB
Ground plane with 5/8 inch adaptor, for AT575-32 antenna above	267-990148-137	SK0044

Table D-4. Recommended Geodetic Active Antennas

The Antenna gain should be selected depending on the cable loss between the antenna and the receiver. Prices and availability can be found in the latest GPS-OEM Price List. You can request this list by sending an e-mail to the GPSPMARKET ( the exact e-mail address is supplied in section 7 of this document ) .

**ACTIVE ANTENNA**

Lower cost antennae for higher volume applications or for more cost sensitive applications are available.

All GPS receivers manufactured by CMC implement a 20dB LNA on board. For this reason, an Active +12dB is more than adequate; antenna with +26 dB and +36 dB may overdrive the RF input of the GPS Receiver, if used with a short cable between the Antenna and the Receiver. The Table D-5 lists the active antennae which could be used with any of the CMC GPS receivers.

Table D-5. Recommended Active Antennae

Typical Applications	CMC Part Numbers	Supplier Part Numbers
AVL (This antenna is currently supplied with ALLSTAR and the SUPERSTAR development kit)	201-990146-716 (MCX connector & 6 meter cable) 201-990146-789 (BNC connector & 6 meter cable) 201-990148-152 (TNC connector & 6 meter cable)	AT575-70W +12 dB
The lowest cost available for AVL	201-990147-432 TNC Female Bulk head	AT575-104W +12dB
Marine application	201-990144-807 TNC Female Bulk head	AT575-68W +12dB

**NOTE:** Prices and availability can be found in the latest GPS-OEM Price List. You can request this list by sending an e-mail to the GPSPMARKET ( the exact e-mail address is supplied in section 7 of this document).

**PASSIVE ANTENNA**

For RTK applications where centimeter-level accuracy is required, it is strongly recommended to use an active geodetic GPS antenna if possible. In the event where the cable length between the receiver and the antenna is very short (less than one meter), a passive antenna could then be considered. The Table D-6 lists the specifications for recommended Passive Antennae patch itself.

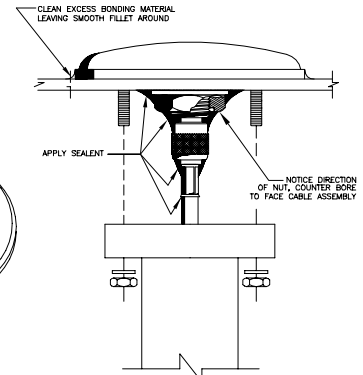
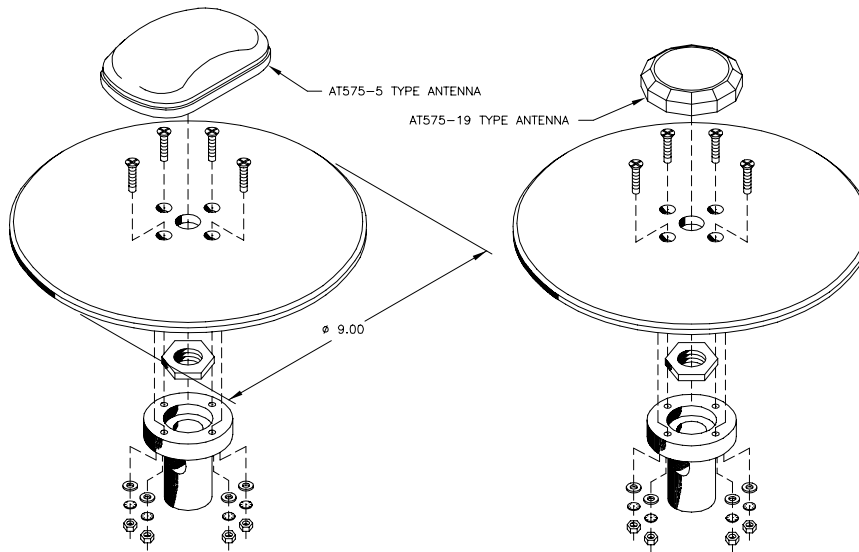
Table D-6. Passive Antenna Specifications (Patch Element)

Frequency	1575 MHz +/- 2 MHz	
Polarization	Right Hand Circular	
Radiation Coverage	4.0 dBic -1.0 dBic -2.5 dBic -4.5 dBic -7.5 dBic	0 degrees 0 < elev. Angle < 75 75 < elev. Angle < 80 80 < elev. Angle < 85 85 < elev. Angle < 90
Connector	TNC Female (most common)	Other connectors also available
Temperature	-55 C to +85C	
Environmental	DO-160C	

Lower cost antennae for higher volume applications or for more cost sensitive applications are available. All GPS receivers manufactured by CMC implement a 20dB LNA on board. For this reason, in many cost sensitive applications it may be necessary to select one of the following passive antennas in conjunction with a low loss coax cable. The Table D-7 lists the passive antennae which could be used with any of the CMC GPS receivers.

Typical Applications	CMC Part Numbers	Supplier Part Numbers
The lowest cost available for AVL	201-990147-433 TNC Female Bulk head	AT575-97CA

Table D-7. Recommended Passive Antennae

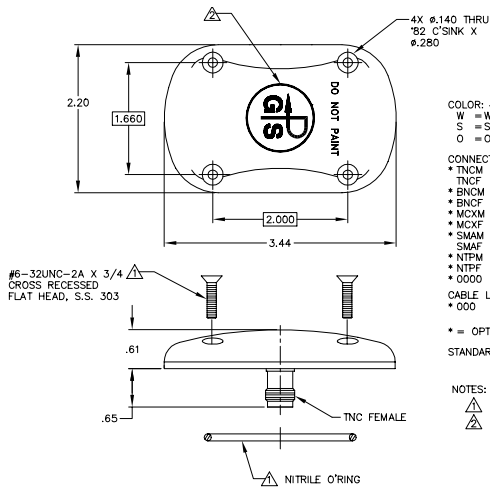


INSTALLATION SUGGESTIONS:

1. THE LOCATION CHOSEN TO MOUNT THE ANTENNA MUST BE A HIGH POINT FREE OF ANY OBSTRUCTION.
2. CHOOSE A LOCATION THAT WILL NOT EXCEED THE RECOMMENDED COAX LENGTH.
3. USE ONLY THE RECOMMENDED CABLE. DO NOT USE ADDITIONAL SPLICES OR CONNECTORS.
4. AVOID LOCATIONS THAT WILL SUBJECT THE ANTENNA TO EXHAUST CONTAMINATION.
5. MOUNT THE ANTENNAS AT LEAST TWO FEET FROM ANY OTHER ANTENNA.
6. IT IS RECOMMENDED TO SEAL THE CONNECTION OF THE CABLE AND ANTENNA. USE SIKAFLEX 221 TYPE SEALER OR EQUIVALENT. APPLY AN ADEQUATE AMOUNT OF SEALER TO CABLE CONNECTOR AFTER IT IS CONNECTED TO THE ANTENNA. COVER THE ANTENNA AND CABLE CONNECTOR WITH SEALER. INSURE LEAK PROOF SEAL. CAUTION: TRIM AWAY ANY EXCESS SEALER AND DO NOT SPREAD SEALER TOWARD BASE MOUNT.
7. CHECK ANTENNA RECEPTION WITH GPS RECEIVER.

9806018

GPS Pre-Amplifier Antenna, AT575-19 Rev E



AT575-32-XXXX-XX-XX-XX

COLOR: W = WHITE  
S = SMOKE GRAY  
O = OLIVE DRAB

CONNECTOR  
\* TNCM = TNC MALE  
\* TNCF = TNC FEMALE  
\* BNCM = BNC MALE  
\* BNCF = BNC FEMALE  
\* MCXM = MCX MALE  
\* MCXF = MCX FEMALE  
\* SMAF = SMA MALE  
\* NTPM = N TYPE MALE  
\* NTPF = N TYPE FEMALE  
\* 0000 = NO TERMINATION  
CABLE LENGTH  
\* 000 = IN INCHES

MAGNET  
\* NM = NO MAGNET  
\* IM = INTERNAL  
\* RM = REMOVABLE  
\* HD = HEAVY DUTY

GAIN (± 2 db)  
00 = PASSIVE  
12 = 12 db (20 mA)  
26 = 26 db (35 mA)  
36 = 36 db (50 mA)  
XX = OTHER

VOLTAGE  
00 = PASSIVE  
05 = 5 VDC  
RG = 5-18 VDC  
XX = OTHER

\* = OPTION NOT AVAILABLE WITH THIS MODEL  
STANDARD MODEL = AT575-32W-TNCF-000-RG-26-NM

NOTES: UNLESS OTHERWISE SPECIFIED.  
△ O-RING AND 4 SCREWS SUPPLIED BY AEROANTENNA TECHNOLOGY.  
△ LOGO OPTIONAL.

SPECIFICATION:

FREQUENCY: 1575 MHz ±10 MHz  
POLARIZATION: RIGHT HAND CIRCULAR  
AXIAL RATIO: 3 dB MAX  
RADIATION COVERAGE: 4.0 dBic 0° = 0°  
-1.0 dBic 0° < θ < 75°  
-2.5 dBic 75° < θ < 80°  
-4.5 dBic 80° < θ < 85°  
-7.5 dBic θ = 90°

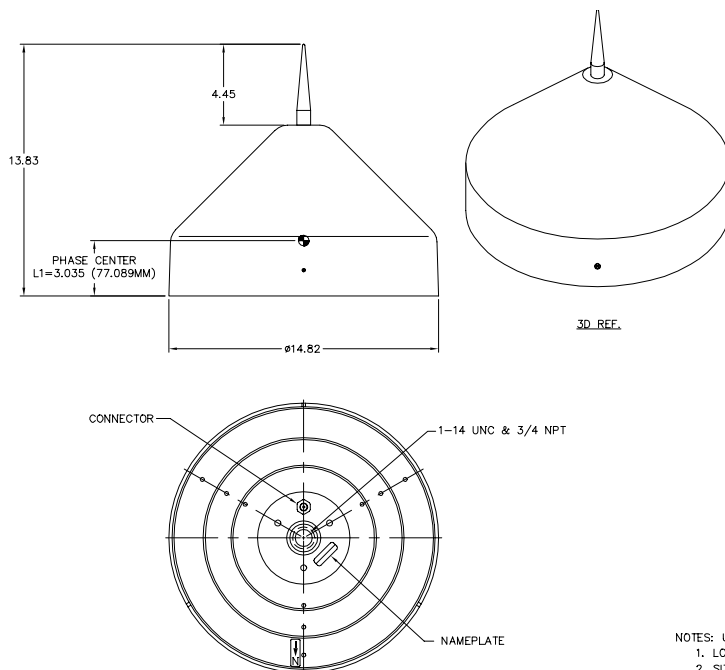
AMPLIFIER:  
NOISE FIGURE: 2.5 dB MAX  
IMPEDANCE: 50 OHM  
VSWR: ≤2.0:1  
BAND REJECTION: 35 dB  
POWER HANDLING: 1 WATT

FINISH: FLUID RESISTANT, POLYURETHANE ENAMEL  
WEIGHT: 3.9 OUNCES  
ALTITUDE: 55,000'  
OPERATING TEMP.: -55°C TO +85°C  
DESIGNED TO: DO-160C,  
ENV CAT: F2-AB(CLY)XSDFDXXXXXXZAC  
ARINC 743  
MIL-STD-810C  
MIL-E-5400 , MIL-C-5541,  
MIL-E-5272C

9804046

CMC PART NUMBER 201-990147-684 (12db)

Outline Drawing GPS Antenna, 1575 MHz, AT575-32 Rev E



**SPECIFICATION:**

FREQUENCY: 575 MHz  $\pm 5$  MHz  
POLARIZATION: RIGHT HAND CIRCULAR  
AXIAL RATIO: 3 dB MAX.  
RADIATION COVERAGE: 4.0 dBic  $\theta' = 0^\circ$   
-1.0 dBic  $0^\circ < \theta < 75^\circ$   
-2.3 dBic  $75^\circ < \theta < 80^\circ$   
-4.5 dBic  $80^\circ < \theta < 85^\circ$   
-7.5 dBic  $\theta = 90^\circ$

AMPLIFIER:  
NOISE FIGURE: 2.5 dB MAX.  
IMPEDANCE: 50 OHM  
VSWR:  $\leq 2.0:1$   
BAND REJECTION: 35 dB  
POWER HANDLING: 1 WATT  
FINISH: POLYURETHANE ENAMEL  
COLOR: WHITE  
WEIGHT: 10 Lbs. MAX.  
ALTITUDE: 20,000'  
OPERATING TEMP.: -55°C TO +85°C  
DESIGNED TO: DO-160C

AT575-90X-XXXX-XXX-XX-XX-XX

COLOR: W = WHITE  
S = SMOKE GRAY  
O = OLIVE DRAB

CONNECTOR  
\*TNCM = TNC MALE  
\*TNCF = TNC FEMALE  
\*BNCM = BNC MALE  
\*BNCF = BNC FEMALE  
\*MCXM = MCX MALE  
\*MCXF = MCX FEMALE  
\*SMAF = SMA FEMALE  
\*SMAF = SMA FEMALE  
\*NTPM = NTP MALE  
\*NTPF = NTP FEMALE  
\*0000 = NO TERMINATION

CABLE LENGTH  
000 = IN INCHES  
\* = OPTION NOT AVAILABLE WITH THIS MODEL  
STANDARD MODEL= AT575-90W-TNCF-000-05-26-NM

MAGNET  
NM = NO MAGNET  
IM = INTERNAL  
RM = REMOVABLE  
HD = HEAVY DUTY

GAIN ( $\pm 2$  db)  
00 = PASSIVE  
12 = 12 DB (20 mA)  
26 = 26 db (35 mA)  
36 = 36 db (50 mA)  
XX = OTHER

VOLTAGE  
00 = PASSIVE  
05 = +5 VDC  
RG = +5 TO +18 VDC  
XX = OTHER

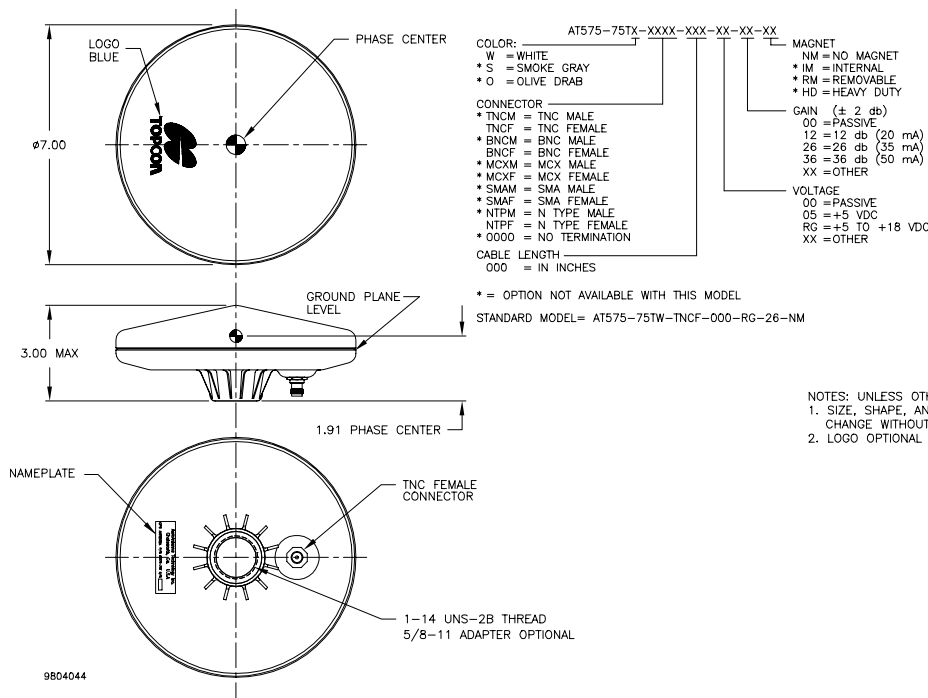
NOTES: UNLESS OTHERWISE SPECIFIED.

1. LOGO OPTIONAL.
2. SIZE, SHAPE, AND CONTENTS OF NAMEPLATE ARE SUBJECT TO CHANGE WITHOUT NOTICE.

CMC PART NUMBER 201-990146-888 (12db)  
CMC PART NUMBER 201-990147-607 (26db)  
CMC PART NUMBER 201-990147-680 (36db)

9804045

Outline Drawing, AT575-90 Rev E



**SPECIFICATION:**

FREQUENCY: 1575 MHz  $\pm 2$  MHz  
POLARIZATION: RIGHT HAND CIRCULAR  $\theta'$   
AXIAL RATIO: 3 dB MAX  
RADIATION COVERAGE: 4.0 dBic  $\theta' = 0^\circ$   
-2.0 dBic  $0^\circ < \theta \leq 70^\circ$   
-4.0 dBic  $70^\circ < \theta \leq 85^\circ$   
-5.0 dBic  $\theta = 90^\circ$

AMPLIFIER:  
NOISE FIGURE: 2.5 dB MAX  
IMPEDANCE: 50 OHM  
VSWR:  $\leq 2.0:1$   
BAND REJECTION: 35 dB @ 1625 MHz  
POWER HANDLING: 1 WATT

MATERIAL: GELOY XP1001  
FINISH: WEATHERABLE POLYMER  
CONNECTOR: SEE TABLE  
WEIGHT: 13 OUNCES MAX  
ALTITUDE: 12,000'  
OPERATING TEMP.: -40°C TO +70°C  
DESIGNED TO: DO-160C

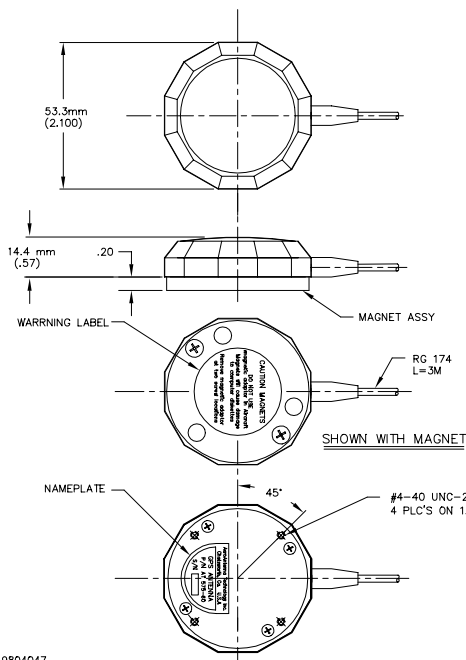
NOTES: UNLESS OTHERWISE SPECIFIED.

1. SIZE, SHAPE, AND CONTENTS OF NAMEPLATE ARE SUBJECT TO CHANGE WITHOUT NOTICE.
2. LOGO OPTIONAL

CMC PART NUMBER 201-990146-887 (12db)  
CMC PART NUMBER 201-990147-606 (26db)  
CMC PART NUMBER 201-990147-679 (36db)

9804044

Outline GPS Antenna, AT575-75T Rev A



AT575-70X-XXXX-XXX-XX-XX-XX

COLOR:  
W = WHITE  
S = SMOKE GRAY  
O = OLIVE DRAB

CONNECTOR:  
TNCM = TNC MALE  
TNCF = TNC FEMALE  
BNCM = BNC MALE  
BNCF = BNC FEMALE  
MCM = MCX MALE  
MCF = MCX FEMALE  
SMA = SMA MALE  
SMF = SMA FEMALE  
NTPM = N TYPE MALE  
NTPF = N TYPE FEMALE  
0000 = NO TERMINATION  
000 = IN INCHES

MAGNET:  
NM = NO MAGNET  
IM = INTERNAL  
RM = REMOVABLE  
HD = HEAVY DUTY

GAIN (± 2 db):  
00 = PASSIVE  
12 = 12 db (20 mA)  
26 = 26 db (35 mA)  
36 = 36 db (50 mA)  
XX = OTHER

VOLTAGE:  
00 = PASSIVE  
05 = 5 VDC  
18 = 18 VDC  
XX = OTHER

\* = OPTION NOT AVAILABLE WITH THIS MODEL.  
STANDARD MODEL = AT575-70W-SMAM-120-05-26-RM  
RIGHT ANGLE CONNECTOR ADD "R" TO CONNECTOR NOTATION.  
THIS OPTION AVAILABLE FOR CABLED ANTENNAS ONLY.

#### SPECIFICATION:

FREQUENCY: 1575 MHz ± 2 MHz  
POLARIZATION: RIGHT HAND CIRCULAR  
AXIAL RATIO: 3 dB MAX  
RADIATION COVERAGE: 4.0 dBic 0° = 0°  
-1.0 dBic 0° < θ < 75°  
-2.5 dBic 75° < θ < 80°  
-4.5 dBic 80° < θ < 85°  
-7.5 dBic θ = 90°

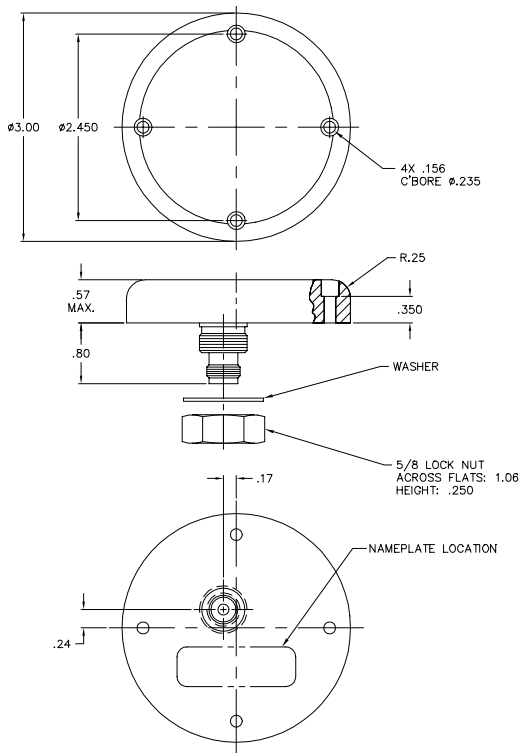
AMPLIFIER:  
GAIN: 22dB MIN  
NOISE FIGURE: 2.5 dB MAX  
VOLTAGE: +3.7 VDC ± 10% BIASED THROUGH CONNECTOR (30 mA MAX)  
IMPEDANCE: 50 OHM  
VSWR: ≤ 2.0:1  
BAND REJECTION: 35 dB  
POWER HANDLING: 1 WATT

FINISH: FLUID RESISTANT, POLYURETHANE ENAMEL  
COLOR: DARK GRAY  
CONNECTOR: SMS FEMALE 522-900-0630H  
WEIGHT: 5 OUNCES  
ALTITUDE: 20,000'  
TEMPERATURE: -20° TO +85°

NOTES: UNLESS OTHERWISE SPECIFIED  
1. LOGO OPTIONAL  
2. SIZE, SHAPE, AND CONTENTS OF NAME PLATE ARE SUBJECT TO CHANGE WITHOUT NOTICE.

CMC PART NUMBER 201-990146-789 (BNCM)  
CMC PART NUMBER 201-990146-7166 (MCXM)

### GPS Pre-Amplifier Antenna, AT575-70 Rev B



#### SPECIFICATION:

FREQUENCY: 1575 MHz ± 10 MHz  
POLARIZATION: RIGHT HAND CIRCULAR  
AXIAL RATIO: 3 dB MAX  
RADIATION COVERAGE: 4.0 dBic 0° = 0°  
+1.0 dBic 0° < θ < 75°  
-2.5 dBic 75° < θ < 80°

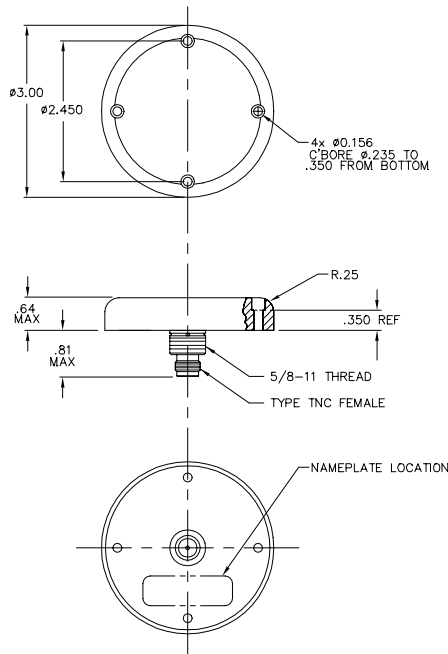
IMPEDANCE: 50 OHM  
VSWR: ≤ 1.5:1  
POWER HANDLING: 1 WATT

FINISH: WEATHERABLE POLYMER, WHITE  
CONNECTOR: TNC FEMALE  
WEIGHT: 89.6g (3.2 OUNCES)  
ALTITUDE: 20,000'  
TEMPERATURE: -55°C TO +85°C  
DESIGNED TO: DO-160C, ENV CAT: F2-AB(CLY)XSFD5XXXXXXL2AC  
MIL-STD-810C, MIL-C-5541, MIL-E-5400, MIL-E-5272C

9806017

### Outline Drawing GPS Antenna, 1575 MHz, AT575-97CA Rev B





9804049

AT575-104X-XXXX-XXX-XX-XX-XX

COLOR: W = WHITE, S = SMOKE GRAY, O = OLIVE DRAB

CONNECTOR: \*TNCM = TNC MALE, \*TNCF = TNC FEMALE, \*BNCM = BNC MALE, \*BNCF = BNC FEMALE, \*MCM = MCX MALE, \*MCMF = MCX FEMALE, \*SMAM = SMA MALE, \*SMAF = SMA FEMALE, \*NTPM = N TYPE MALE, \*NTPF = N TYPE FEMALE, \*0000 = NO TERMINATION

CABLE LENGTH: \*000 = IN INCHES

\* = OPTION NOT AVAILABLE WITH THIS MODEL

STANDARD MODEL = AT575-1040-TNCF-000-05-26-NM

RIGHT ANGLE CONNECTOR ADD "R" TO CONNECTOR NOTATION. THIS OPTION AVAILABLE FOR CABLED ANTENNAS ONLY.

1. 5/8 LOCK WASHER AND NUT SUPPLIED. INSTALLATION TORQUE: 5 FT LB MAX.

NOTES: UNLESS OTHERWISE SPECIFIED.

MAGNET: \*NM = NO MAGNET, \*IM = INTERNAL, \*RM = REMOVABLE, \*HD = HEAVY DUTY

GAIN ( $\pm 2$  db): 00 = PASSIVE, 12 = 12 db (20 mA), 26 = 26 db (35 mA), 36 = 36 db (50 mA), XX = OTHER

VOLTAGE: 00 = PASSIVE, 05 = 5 VDC, \*RG = 5-18 VDC, XX = OTHER

**SPECIFICATION:**

FREQUENCY: 1575 MHz  $\pm 10$  MHz

POLARIZATION: RIGHT HAND CIRCULAR

AXIAL RATIO: 3 dB MAX

RADIATION COVERAGE: 4.0 dBic  $0^\circ = 0^\circ$ , -1.0 dBic  $0^\circ < \theta < 75^\circ$ , -2.5 dBic  $75^\circ < \theta < 80^\circ$ , -4.5 dBic  $80^\circ < \theta < 85^\circ$ , -7.5 dBic  $\theta = 90^\circ$

AMPLIFIER: GAIN: SPECIFY WHEN ORDERING

NOISE FIGURE: 2.5 dB MAX

IMPEDANCE: 50 OHM

VSWR:  $\leq 2.0:1$

BAND REJECTION: 35 dB

POWER HANDLING: 1 WATT

FINISH: WEATHERABLE POLYMER, MATT TEXTURE

CONNECTOR: SPECIFY WHEN ORDERING

WEIGHT: 4 OUNCES

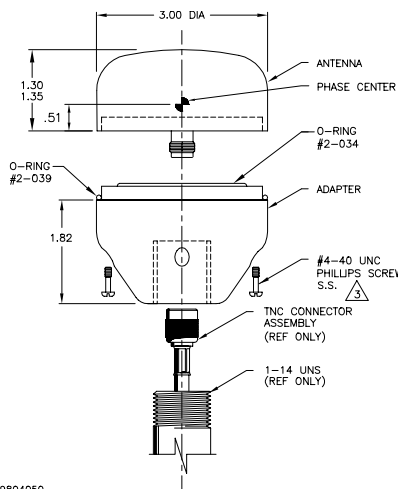
ALTITUDE: 55,000'

TEMPERATURE: -55°C TO +85°C

DESIGNED TO: DO-160C, ENV CAT: F2-AB(CLY)XSFDXXXXXX2AC, ARINC 743, MIL-STD-883C, MIL-C-5541, MIL-E-5400, MIL-E-5272C

CMC PART NUMBER 201-990147-432

## Outline Drawing GPS Antenna, 1575 MHz, AT575-104 Rev -



9804050

AT575-68X-XXXX-XXX-XX-XX-XX

COLOR: W = WHITE, S = SMOKE GRAY, O = OLIVE DRAB

CONNECTOR: \*TNCM = TNC MALE, \*TNCF = TNC FEMALE, \*BNCM = BNC MALE, \*BNCF = BNC FEMALE, \*MCM = MCX MALE, \*MCMF = MCX FEMALE, \*SMAM = SMA MALE, \*SMAF = SMA FEMALE, \*NTPM = N TYPE MALE, \*NTPF = N TYPE FEMALE, \*0000 = NO TERMINATION

CABLE LENGTH: \*000 = IN INCHES

\* = OPTION NOT AVAILABLE WITH THIS MODEL

STANDARD MODEL = AT575-68W-TNCF-000-05-26-NM

RIGHT ANGLE CONNECTOR ADD "R" TO CONNECTOR NOTATION. THIS OPTION AVAILABLE FOR CABLED ANTENNAS ONLY.

NOTES: UNLESS OTHERWISE SPECIFIED.

1. LOGO OPTIONAL.

2. SIZE, SHAPE, AND CONTENTS OF NAMEPLATE ARE SUBJECT TO CHANGE WITHOUT NOTICE.

ASSEMBLE ANTENNA TO ADAPTER USING 4 SCREWS ON CROSS MOTION. DO NOT OVER TORQUE THE SCREWS!

MAGNET: \*NM = NO MAGNET, \*IM = INTERNAL, \*RM = REMOVABLE, \*HD = HEAVY DUTY

GAIN ( $\pm 2$  db): 00 = PASSIVE, 12 = 12 db (20 mA), 26 = 26 db (35 mA), 36 = 36 db (50 mA), XX = OTHER

VOLTAGE: 00 = PASSIVE, 05 = 5 VDC, RG = 5-18 VDC, XX = OTHER

**SPECIFICATION:**

FREQUENCY: 1575 MHz  $\pm 10$  MHz

POLARIZATION: RIGHT HAND CIRCULAR

AXIAL RATIO: 3 dB MAX

RADIATION COVERAGE: 4.0 dBic  $0 = 0^\circ$ , -1.0 dBic  $0 < \theta = 75^\circ$ , -3.0 dBic  $\theta = 90^\circ$

VSWR:  $\leq 1.5:1$

IMPEDANCE: 50 OHM

POWER HANDLING: 1 WATT

FINISH: OUTDOOR-WEATHERABLE POLYMER

WEIGHT: 8oz

ALTITUDE: 20,000'

TEMPERATURE: -55°C TO +85°C

DESIGNED TO: DO-160C, MIL-C-5541, MIL-E-5272C

CMC PART NUMBER 201-990144-807 (12db)

CMC PART NUMBER 201-990144-589 (0db)

CMC PART NUMBER 201-990144-791 (26db)

## GPS Pre-Amplifier Antenna, AT575-68 Rev F

4
3
2
1

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NOTES:

1. PART NO. 217-601730-XXXX, LAST 2 DIGITS = LENGTH OF CABLE IN METER

2. CABLE CONSTRUCTION:

CENTER CONDUCTOR	#20 BARE COPPER WIRE, RESISTANCE - 33.1 OHMS PER KM.
INSULATION	POLYETHYLENE
INNER SHIELD	ALUMINUM FOIL - 100% COVERAGE
OUTER SHIELD	TINNED COPPER BRAID - 85% COVERAGE, RESISTANCE 45.9 OHMS PER KM.
JACKET	BLACK PVC
NOMINAL IMPEDANCE	50 OHMS
NOMINAL VEL. OF PROPAGATION	66%
NOMINAL CAPACITY	101.7 PF. PER METER
ATTENUATION	@ 1000MHZ 44.3DB PER 100 METERS

REVISION STATUS OF SHEETS											REVISIONS				
SHEET	1	2	3	4	5	6	7	8	9	10	ZONE	REV	DESCRIPTION	DATE	APPROVED
REV	-											-	PRODUCT BASELINE RA 98-0004	98-05-06	D.C.

A LAS 2 D G S OF PAP NO. N MC CP

TABLE		
PART NUMBER	CONNECTOR 1	CONNECTOR 2
217-601730-001 THRU -099	TNC MALE STRAIGHT	TNC MALE STRAIGHT
217-601730-100 THRU -199	TNC MALE STRAIGHT	TNC MALE RIGHT ANGLE
217-601730-200 THRU -299	TNC MALE RIGHT ANGLE	TNC MALE RIGHT ANGLE
217-601730-300 THRU -399	BNC MALE STRAIGHT	TNC MALE STRAIGHT
217-601730-400 THRU -499	BNC MALE STRAIGHT	TNC MALE RIGHT ANGLE
217-601730-500 THRU -599	BNC MALE STRAIGHT	BNC MALE STRAIGHT

THIRD ANGLE PROJECTION

ANGULAR TOLERANCE

--- ALLSTAR

NEXT ASSY USED ON

APPLICATIONS

CNC REF ONLY

UNLESS OTHERWISE SPECIFIED: DIMENSIONS APPLY AFTER FINISHING AND ARE IN

INCHES	TOLERANCES	MILLIMETERS
.XX .XXX	.XX .XXX	± .6

MATERIAL

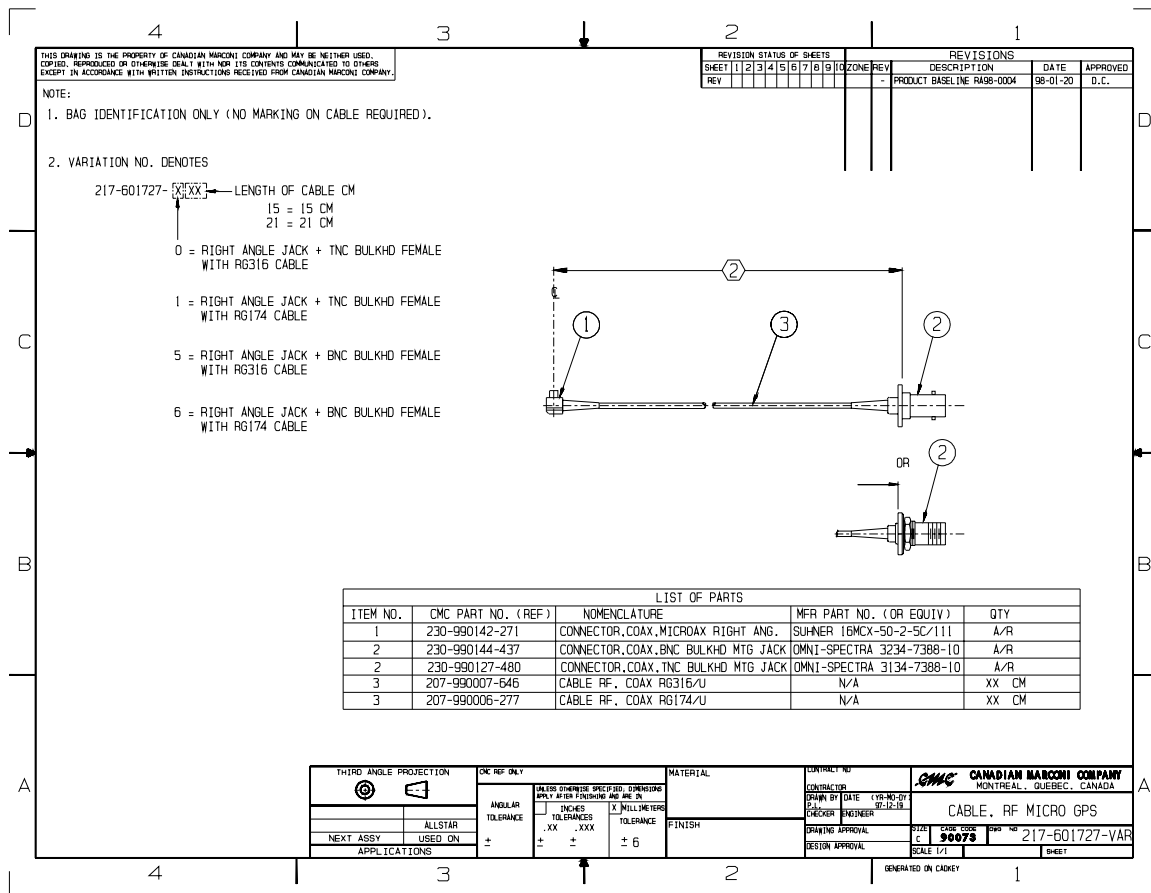
FINISH

SEE SEPARATE PARTS LIST

4
3
2
1

CONTRACT NO.		CANADIAN MARCONI COMPANY	
CONTRACTOR		MONTREAL, QUEBEC, CANADA	
DRAWN BY DATE	100-NO-DT		
D.C.	98-02-13		
CHECKED	ENGINEER		
D.C.	D.GERMAIN		
DRAWING APPROVAL		SIZE	CAGE CODE
H. PELLERIN		C	90073
DESIGN APPROVAL		DWG NO	217-601730-VAR
SCALE 1/1		SHEET	

GENERATED ON PROTEUS



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## APPENDIX E - SUPPORTED DATUM LIST

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DATUM DESCRIPTION TABLE .....	E-1
ELLIPSOID DESCRIPTION TABLE .....	E-4

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## APPENDIX E

### SUPPORTED DATUM LIST

**DATUM DESCRIPTION TABLE**

#	NAME	ELLIPSE	DX	DY	DZ	COUNTRIES
0	WGS 1984	WGS-84	0	0	0	Global definition
1	User Defined 1					
2	User Defined 2					
3	Adindan	Clarke_1880	-161	-14	205	Sudan
4	Arc 1950	Clarke_1880	-143	-90	-294	Botswana, Lesotho, Malawi, etc.
5	Arc 1950	Clarke_1880	-169	-19	-278	Zaire
6	Arc 1960	Clarke_1880	-160	-6	-302	Kenya, Tanzania
7	Australian Geodetic 1984	Australian_National	-134	-48	149	Australia, Tasmania
8	Bogota Observatory	International	307	304	-318	Colombia
9	Campo Inchauspe	International	-148	136	90	Argentina
10	Cape	Clarke_1880	-136	-108	-292	South Africa
11	Carthage	Clarke_1880	-263	6	431	Tunisia
12	Chatham Island Astro 1971	International	175	-38	113	New Zealand (Chatham Island)
13	Chua Astro	International	-134	229	-29	Paraguay
14	Corrego Alegre	International	-206	172	-6	Brazil
15	European 1950	International	-87	-98	-121	Austria, Belgium, Denmark, Finland, France, West Germany, Gibraltar, Greece, Italy, Luxembourg, Netherlands, Norway, Portugal, Spain, Sweden, Switzerland
16	European 1950	International	-104	-101	-140	Cyprus
17	European 1950	International	-130	-117	-151	Egypt
18	European 1950	International	-86	-96	-120	England, Channel Islands, Ireland, Scotland, Shetland Islands
19	European 1950	International	-117	-132	-164	Iran
20	European 1950	International	-97	-88	-135	Italy (Sicily)
21	European 1979	International	-86	-98	-119	Austria, Finland, Netherlands, Norway, Spain, Sweden, Switzerland
22	Geodetic Datum 1949	International	84	-22	209	New Zealand
23	Hjorsey 1955	International	-73	46	-86	Iceland

#	NAME	ELLIPSE	DX	DY	DZ	COUNTRIES
24	Hong Kong 1963	International	-156	-271	-189	Hong Kong
25	Hu-Tzu-Shan	International	-637	-549	-203	Taiwan
26	Indian 1954	Everest 1830	218	816	297	Thailand, Vietnam
27	Ireland 1965	Airy_modified	506	-122	611	Ireland
28	Kertau 1948	Everest 1948	-11	851	5	West Malaysia & Singapore
29	Liberia 1964	Clarke_1880	-90	40	88	Liberia
30	Luzon	Clarke_1866	-133	-77	-51	Philippines (Excluding Mindanao)
31	Massawa	Bessel_1841	639	405	60	Ethiopia (Eritrea)
32	Merchich	Clarke_1880	31	146	47	Morocco
33	Minna	Clarke_1880	-92	-93	122	Nigeria
34	Nahrwan	Clarke_1880	-247	-148	369	Oman (Masirah Island)
35	North American 1927	Clarke_1866	-5	135	172	Alaska
36	North American 1927	Clarke_1866	-3	142	183	Antigua, Barbados, Bermuda, Caicos Islands, Cuba, Dominican Republic, Grand Cayman, Jamaica, Turks Islands
37	North American 1927	Clarke_1866	-10	158	187	Canada
38	North American 1927	Clarke_1866	-7	162	188	Canada (Alberta, British Columbia)
39	North American 1927	Clarke_1866	-9	157	184	Canada (Manitoba, Ontario)
40	North American 1927	Clarke_1866	-22	160	190	Canada (New Brunswick, Newfoundland, Nova Scotia, Quebec)
41	North American 1927	Clarke_1866	4	159	188	Canada (Northwest Territories, Saskatchewan)
42	North American 1927	Clarke_1866	-7	139	181	Canada (Yukon)
43	North American 1927	Clarke_1866	0	125	201	Canal zone
44	North American 1927	Clarke_1866	0	125	194	Central America
45	North American 1927	Clarke_1866	-12	130	190	Mexico
46	North American 1983	GRS-80	0	0	0	Alaska, Canada, CONUS, Central America, Mexico
47	Old Egyptian 1907	Helmert_1906	-130	110	-13	Egypt
48	Old Hawaiian	Clarke_1866	61	-285	-181	Hawaii, Kauai, Maui, Oahu
49	Oman	Clarke_1880	-346	-1	224	Oman
50	Ord. Survey G. Britain 1936	Airy	375	-111	431	England, Isle of Man, Scotland, Shetland Islands, Wales
51	Pitcairn Astro 1967	International	185	165	42	Pitcairn Island
52	Qatar National	International	-128	-283	22	Qatar
53	Qornoq	International	164	138	-189	Greenland (South)



#	NAME	ELLIPSE	DX	DY	DZ	COUNTRIES
54	Schwarzeck	Bessel_1841_in_Namibia	616	97	-251	Namibia
55	South American 1969	South_America_1969	-57	1	-41	Argentina, Bolivia, Brazil, Chile, Colombia, Ecuador, Guyana, Paraguay, Peru, Trinidad & Tobago, Venezuela
56	South American 1969	South_America_1969	-60	-2	-41	Brazil
57	South American 1969	South_America_1969	-44	6	-36	Colombia
58	South American 1969	South_America_1969	-45	8	-33	Venezuela
59	South Asia	Modified Fisher 1960	7	-10	-26	Singapore
60	Tananarive Observatory 1925	International	-189	-242	-91	Madagascar
61	Tokyo	Bessel_1841	-148	507	685	Japan
62	Tokyo	Bessel_1841	-128	481	664	Mean Value
63	WGS 1972	WGS-72	0	0	0	Global definition

**ELLIPSOID DESCRIPTION TABLE**

Ellipsoid name	Semi-major axis (a)	Inverse flattenning (1/f)
Airy	6377563.3960	299.324964600
Airy_modified	6377340.1890	299.324964600
Australian_National	6378160.0000	298.250000000
Bessel 1841	6377397.1550	299.152812800
Bessel 1841 in Namibia	6377483.8650	299.152812800
Clarke 1866	6378206.4000	294.978698200
Clarke 1880	6378249.1450	293.465000000
Everest (Sabah & Sarawak)	6377298.5560	300.801700000
Everest 1830	6377276.3450	300.801700000
Everest 1948	6377304.0630	300.801700000
Everest 1956	6377301.2430	300.801700000
Everest_Modified	6377304.0630	300.801700000
GRS-80	6378137.0000	298.257222101
Helmert 1906	6378200.0000	298.300000000
Hough	6378270.0000	297.000000000
International	6378388.0000	297.000000000
Krassovsky	6378245.0000	298.300000000
Modified Fisher 1960	6378155.0000	298.300000000
SGS 85	6378136.0000	298.257000000
South America 1969	6378160.0000	298.250000000
WGS-72	6378135.0000	298.260000000
WGS-84	6378137.0000	298.257223563

## APPENDIX F - SOFTWARE REPROGRAMMING MODE

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## APPENDIX F

### SOFTWARE REPROGRAMMING MODE

#### GENERAL

The receiver has an integrated reprogramming facility. The receiver has Flash Memory devices that allow software updates via the RS-232 communication port in less than 2 minutes. The OEM circuit card assembly is forced to enter reprogramming mode when the voltage at the input pin DISC\_IP\_1 is HI. More information on this mode is available on request.

#### PROGRAMMING MODE PROCEDURE

The receiver can be set to programming mode by hardware or by software.

##### 1. PROGRAMMING MODE SETTING BY HARDWARE

- a. At the 26 pin ZIF connector (J1), tie pin 9 to the 5V supply.
- b. At the 20 pin header connector (J3), tie pin 6 to the 5V supply
- c. Apply either 5V supply voltage to the receiver or a master reset pulse to the power control input pin (J1-5 or J3-3)

Notes:

1. If a programming adapter (CMC #220-600932-000) is used, set S2 to PROG then apply 5V supply voltage or press S1 if supply voltage is already applied.
2. The baud rate for programming mode setting by hardware is 19200.

##### 2. PROGRAMMING MODE SETTING BY SOFTWARE

To set the programming mode by software, enter message \$PMCAG,006 when in NMEA mode or message ID #112 when in Binary mode. In either case the receiver will be forced to enter the programming mode at a specific baud rate.

The commands are sent at the operating baud rate. Once these commands are decoded, the receiver will enter programming mode at the specified baud rate, independently of the operating baud rate.

<b>Example</b>	<b>NMEA : \$PMCAG,006,19.2*7A&lt;CR&gt;&lt;LF&gt;</b>
	<b>CMC Binary : 0x01,0x70,0x8F,0x01,0x40,0x41,0x01</b>

The programming utility will send one of these commands to force the programming mode, thus avoiding the need to tie the DISC\_IP\_1 pin to 5 volts. It is useful when the system does not provide external access to the DISC\_IP\_1 pin.

## HOW TO VERIFY IF IN PROGRAMMING MODE OR NOT

Once in programming mode, the receiver sends the following information to both communication ports :

**Ready !!!  
wwwww**

Character "**w**" means waiting for data exchange and will be repeated until the programming utility starts to send data.

The baud rate will be 19200 if the programming mode setting is done by hardware and any other baud rate if done through operational software command.

## WHICH PORT TO USE

The MAIN port or the AUXILIARY port can be used to program the receiver. It is recommended to stop any communication on the unused port for proper operation in programming mode.

## PROGRAMMING UTILITY

The programming utility "PROG.EXE" is used to :

- a. Set the receiver to programming mode (if not already done)
- b. Erase the Operational S/W
- c. Transfer the new operational S/W data to the receiver
- d. Verify if the operation has been done successfully

Type "PROG" at the DOS prompt to get help information on the utility :

Example:

C:\>PROG

PROGRAMMING UTILITY VERSION : 1.104

NOTE : this utility uses the serial port interrupt

Example:

PROG UGPSO.SUM 1 0 1 0 <ENTER>

parameter 1 : Operational S/W filename  
parameter 2 : PC Serial Port (1or2)  
parameter 3 : 0:CMC Binary 1:NMEA protocol  
parameter 4 : Synchronisation baud rate (300 to 19200)  
parameter 5 : Data transfer baud rate (300 to 38400)

Baud Rate Code :

0:38400 1:19200 2:9600 3:4800 4:2400 5:1200 6:600 7:300

Parameters 3, 4 and 5 are optional and their default values are: 0 1 0

## PROG.EXE PARAMETERS

The PROG.EXE utility requires the following parameters:

**PROG** **FILENAME**, **COM#** , **MODE** **BAUD\_RATE**, **TRANSFER\_B\_R**

Parameters **MODE**, **BAUD\_RATE** and **TRANSFER\_B\_R** are optional and have the following default values:

**FILENAME** : New receiver binary file (provided by CMC)

**COM#**: PC Serial Communication Port presently used

COM 1: 1

COM 2: 2

**MODE** : Actual receiver operating mode

CMC Binary: 0

NMEA : 1

Default Value: 0 (CMC Binary)

**SYNC\_B\_R**: Synchronization Baud Rate

38400: 0

19200: 1

9600: 2

4800: 3

2400: 4

1200: 5

600: 6

300: 7

Default Value: 1 (19200)

Note: This baud rate will be used by the receiver to start data exchange with the Programming utility when in Programming mode.

**TRANSFER\_B\_R** : Data Transfer Baud Rate (see **SYNC\_B\_R** description for possible values)

Default Value: 0 (38400)

**PROGRAMMING UTILITY ALGORITHM**

1. Programming utility performs the following tasks:
  - a. Verify the parameter values
  - b. Verify the receiver binary file validity
  - c. Send the input message \$PMCAG,006 (MODE = 1) or message #112 (MODE =0) at the specified baud rate (SYNC\_B\_R parameter)
  - d. Wait for "w" character (@ SYNC\_B\_R)
  - e. Start & Verify Erase process
  - f. Change communication baud rate to the TRANSFER\_B\_R value
  - g. Start and Verify Programming process
2. Examples
  - a. If the receiver is forced into programming mode via the DISC\_IP\_1 input pin, only the following command is necessary :  
  
                  PROG FILENAME 1 (if COM 1 in use)  
                                      (see Default Value for other parameters)
  - b. If the receiver is in NMEA mode @ 9600 :  
  
                  PROG FILENAME 1 1 2 0  
  
                  This forces the utility to send NMEA message \$PMCAG,006  
                  on COM 1 @ 9600 and to transfer the binary data @ 38400.
  - c. If the receiver is in CMC Binary mode @ 19200 and wish to set the transfer baud rate @ 19200 :  
  
                  PROG FILENAME 1 0 1 1
  - d. If the receiver sends "w" characters @ 4800 :  
  
                  PROG FILENAME 1 0 3 0

In programming mode, the MODE parameter is no longer important and can be set to "0" or "1". But the SYNC\_B\_R parameter has to be set to the programming mode baud rate in use by the receiver (sending "w" characters) .



**930-600021-000**